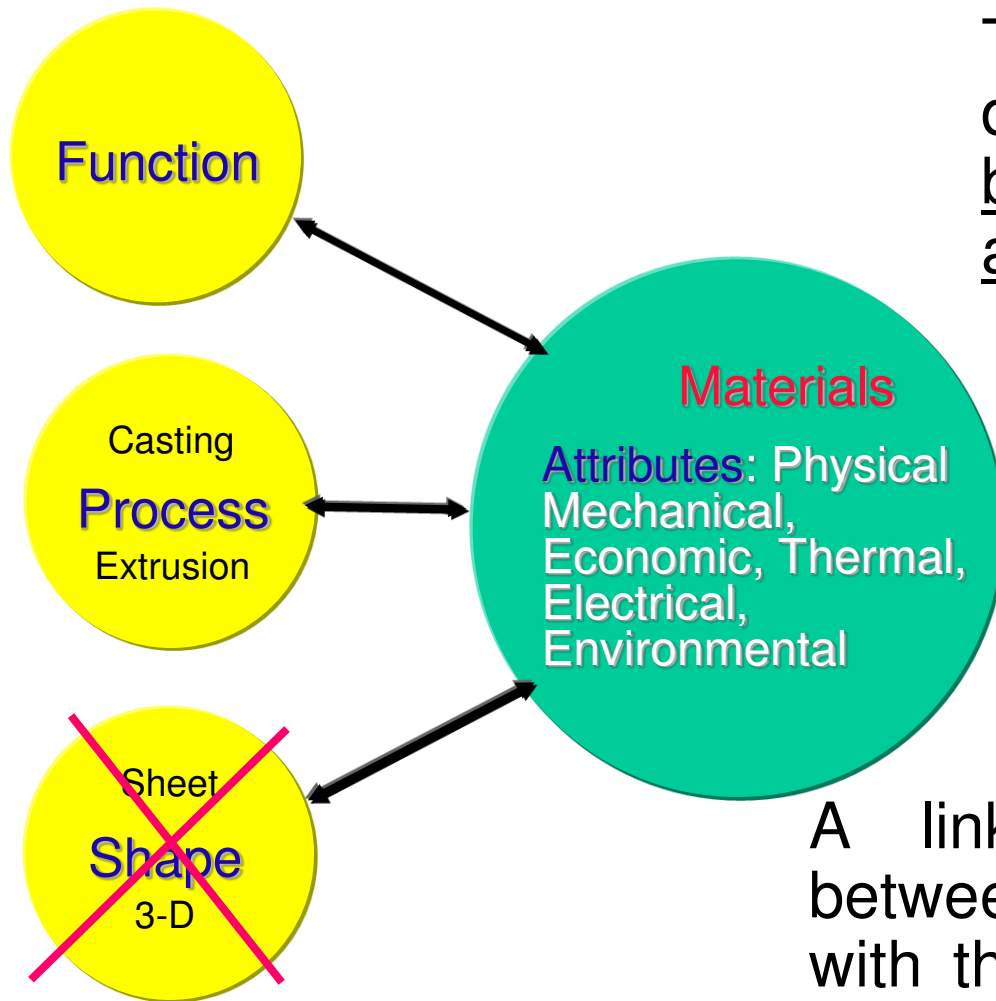


## Chapter Three (Part 1)



## Materials Performance Indices (Without Shape)

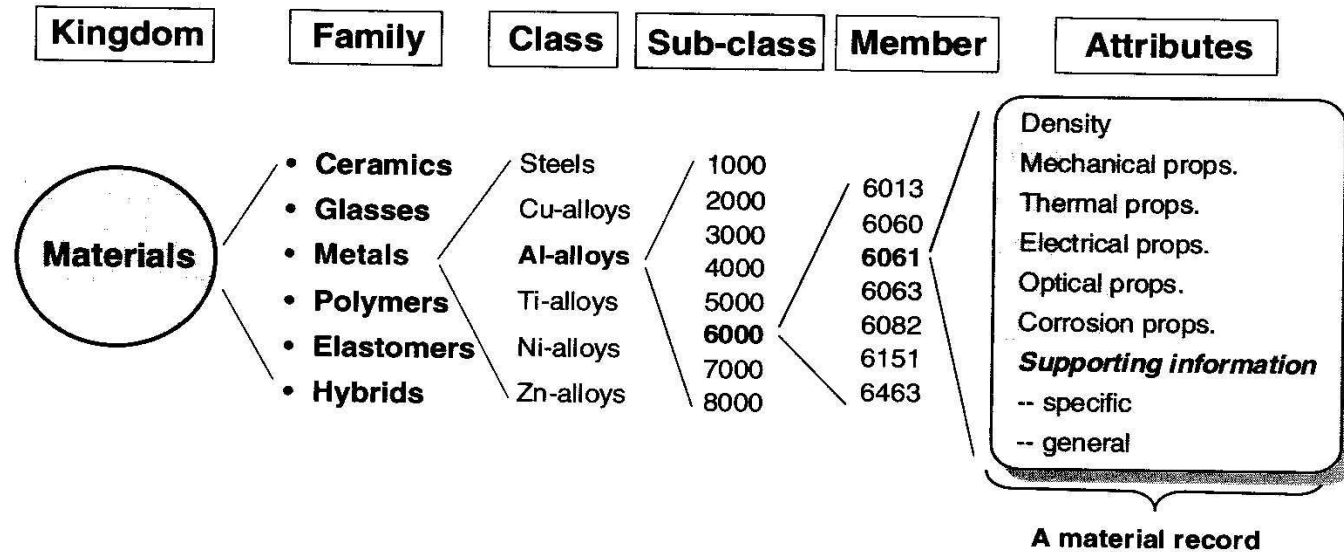
The selection of a material depends on the interaction between the MATERIAL and FUNCTION.



Adapted from M.F. Ashby

A link must be established between MATERIAL, FUNCTION with the PROCESS and SHAPE playing an important role

# MATERIAL ATTRIBUTES



The taxonomy of the kingdom of materials and their attributes. Computer-based selection software stores data in a hierarchical structure like this.

- An engineering component has:  
(boundary condition for Materials Selection)
- 1. **Function**: to carry load, transmit heat, contain a pressure, etc..  
*What does the component do?*
- 2. **Objectives**: as cheap as possible, light, safe, strong, etc...  
*What is to be Maximised or Minimised?*
- 3. **Constraints**: subject to constraints such as carry load without failure, certain dimensions are fixed, cost is within limits etc...
  - *What non-negotiable conditions are to be met? (Rigid)*
  - *What negotiable but desirable conditions? (Soft)*

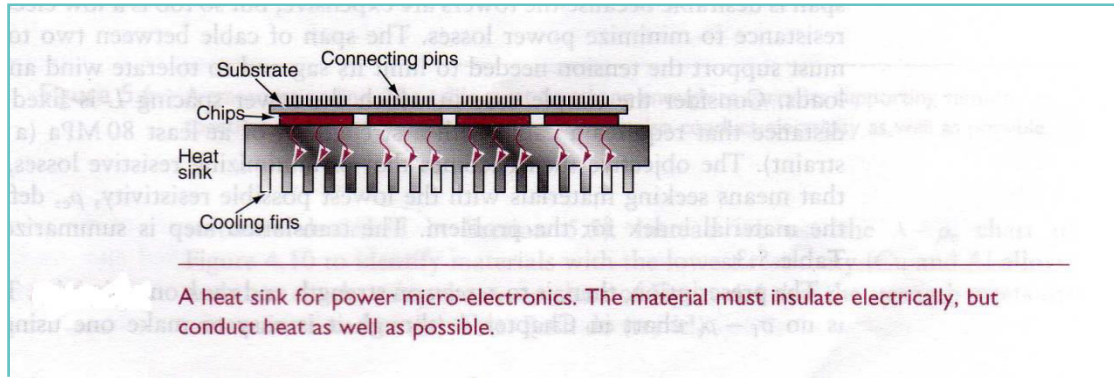


4. **Free Variables**: materials choice, cross-section area, thickness, and length are free

Which design variables are free? (variables which can be changed)

e.g. an engineering component has:  
(boundary condition for Materials Selection)

## Heat Sink for hot microchips



Function

Objective

Constraints

Free variables

Heat sink

Maximize thermal conductivity

- Materials must be good insulator

- All dimensions are specified

Choice of material

- Two concepts are used in the selection procedure:

### **1. Materials Performance Index**

Combination of materials properties that characterise the performance of a material in a given application (Ashby)

### **2. Materials Selection Charts**

Plots of materials properties that form the maximising factors

## Two concepts are used in the selection procedure:

### 1) Material Performance Indices

- Performance of a component/structure is specified by:
  1. *Functional requirements (function) (F) e.g. carry loads, transmit energy, store energy etc.*
  2. *Geometry, (G)*
  3. *Materials properties, (M)*

Performance:

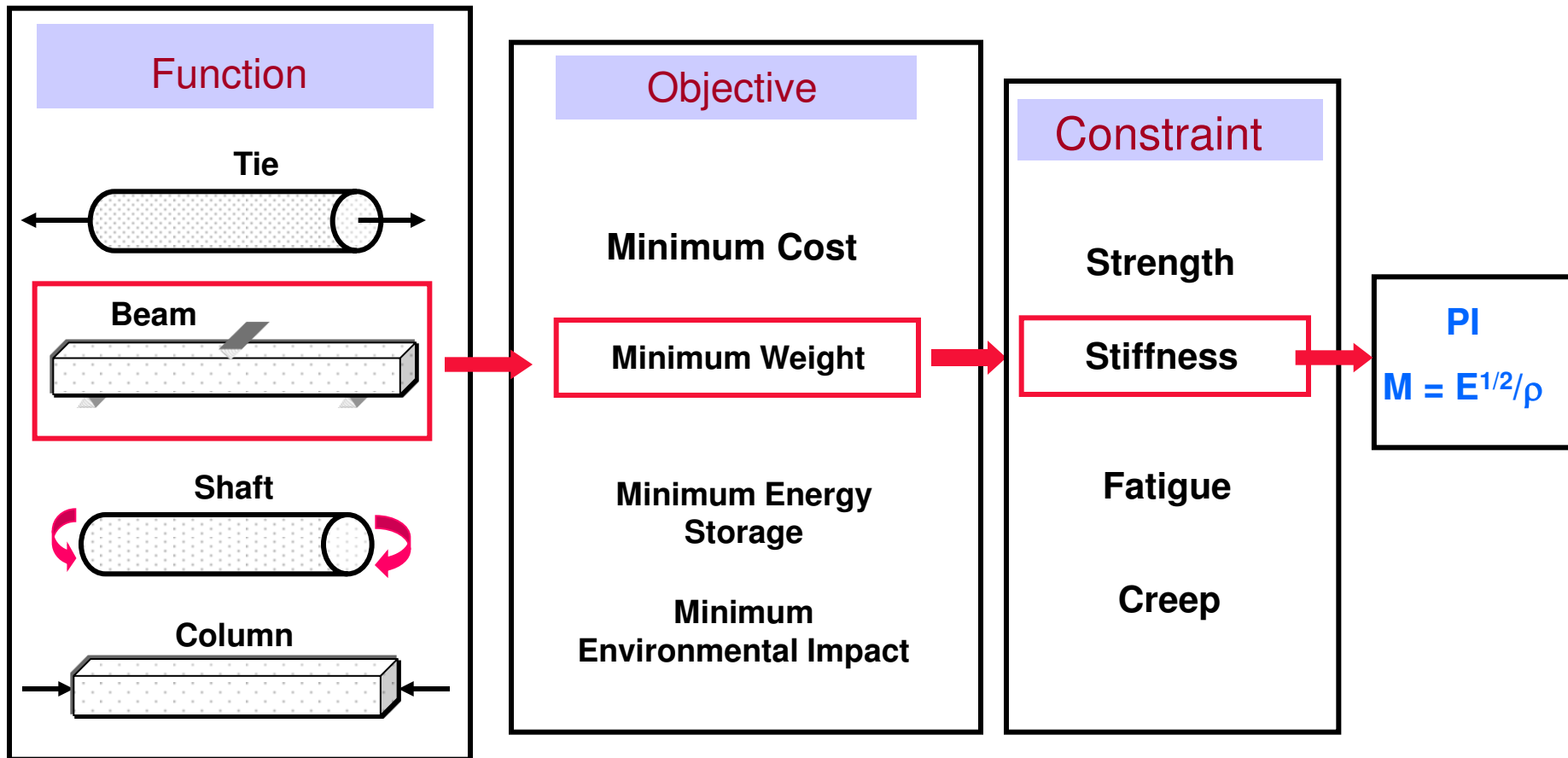
$P = f[(\text{functional requirements, } F); (\text{geometry, } G); (\text{materials properties, } M)]$

$$P = f(F, G, M)$$

- Optimum design is selection of a material which **maximise** (strength, stiffness) or **minimise** (weight, cost) the performance, P.
- In many cases, the function, geometry and materials properties are independent of each other and are said to be separable.

$$P = f_1(F) \times f_2(G) \times f_3(M)$$

- Which means that the optimum selection of a material is independent of the details of the design. It is the same for all geometries and values of functional requirements.
- Performance is maximised by maximising  $f_3(M)$  and is called *“Performance Index, M”*



Specification of *Function*, *Objective* and *Constraint* leads to a material performance index M. (after Ashby, 1999)



---

**Functions and Constraints**

---

**Maximise (Stiffness)****Tie (tensile strut)**

*Stiffness, length specified, section area free*

$$E/\rho$$

**Beam (loaded in bending)**

*Stiffness, length, shape specified, section area free*

$$E^{1/2}/\rho$$

*Stiffness, length, height specified, width free*

$$E/\rho$$

*Stiffness, length, width specified, height free*

$$E^{1/3}/\rho$$

**Panel (flat plate, loaded in bending)**

*Stiffness, length, width specified, thickness free*

$$E^{1/3}/\rho$$

**Panel (flat plate, buckling failure)**

*Collapse load, length and width specified, thickness free*

$$E^{1/3}/\rho$$

**After M.F. Ashby**

Strength (not  
stiffness)

---

## Functions and Constraints

---

Maximise (Strength)

### **Tie (tensile strut)**

*Stiffness, length specified, section area free*

$$\sigma_f / \rho$$

### **Beam (loaded in bending)**

*Stiffness, length, shape specified, section area free*

$$\sigma_f^{2/3} / \rho$$

*Stiffness, length, height specified, width free*

$$\sigma_f / \rho$$

*Stiffness, length, width specified, height free*

$$\sigma_f^{1/2} / \rho$$

### **Panel (flat plate, loaded in bending)**

*Stiffness, length, width specified, thickness free*

$$\sigma_f^{1/2} / \rho$$

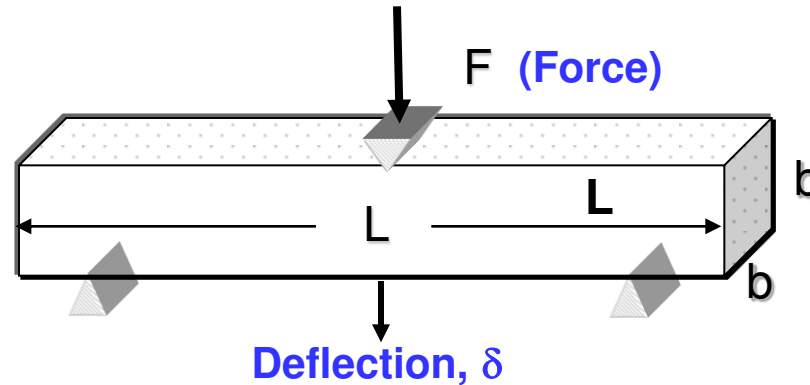
### **Panel (flat plate, buckling failure)**

*Collapse load, length and width specified, thickness free*

$$\sigma_f^{1/2} / \rho$$

After M.F. Ashby

## Example 1: Performance Index for a light, stiff beam



- **FUNCTION :** Beam (bending)
- **OBJECTIVE :** minimise mass (weight)
- **CONSTRAINT(S):** Stiffness is  $S = F/\delta$  (must not deflect more than  $\delta$ )
- **FREE VARIABLE(S):** Cross-sectional area ( $A$ ), material

- The beam of square section, loaded in bending. Its stiffness is  $S = F/\delta$  (must not deflect more than  $\delta$ )

- Constraint on stiffness is given by:

$$S = \frac{F}{\delta} \geq \frac{C_1 EI}{l^3} \quad (1)$$

- For a square cross section, “ $I$ ” is given by:

$$I = \frac{b^4}{12} = \frac{A^2}{12} \quad (2)$$

- The mass (**Objective**),  $m$ , is given by:

$$m = Al\rho \quad (3)$$

- The free variable is the cross section area,  $A = b^2$
- Combining equations (1), (2) and (3), the free variable,  $A$ , can be eliminated and we then obtain:

$$m \geq \left( \frac{12S}{C_1 l} \right)^{1/2} l^3 \left( \frac{\rho}{E^{1/2}} \right) \quad (4)$$

$F$     $G$     $M$     $\longrightarrow$    PERFORMANCE INDEX

The best materials for a light, stiff beam are those with the smallest value of  $\rho/E^{1/2}$  .  
Therefore, to minimize the mass

$$M = \frac{E^{1/2}}{\rho}$$

PERFORMANCE INDEX

**OBJECTIVE : minimize mass (weight)**



## Example 1: Performance Index for a light, stiff beam

- A **light** and **stiff** beam (*cross section free*) is one with the largest:

$$M = \frac{E^{1/2}}{\rho}$$

- If *height is free*, the material index is:

$$M = \frac{E^{1/3}}{\rho}$$

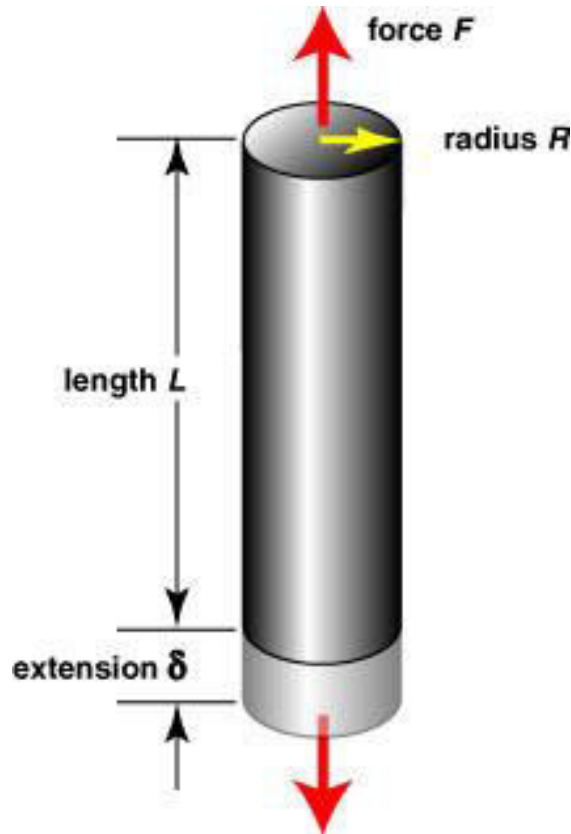
- If the *width is free*, the material index is:

$$M = \frac{E}{\rho}$$

## Steps in deriving a “performance Index”

1. Identify the primary **FUNCTION**
2. Develop an equation for the **OBJECTIVE** (objective function): e.g; weight, cost, etc... (to be maximised or minimised). Objective function contains one or more free variables.
3. Identify the **CONSTRAINTS** (which must be met), rank them in order of importance
4. Identify the **FREE VARIABLES** (unspecified)
5. Develop equations for the constraints (no failure, no buckling, cost below target...)
6. Eliminate the free variable(s) in the objective equation using the constraints.
7. Group the variables into 3 groups: F, G, M
8. Read-off the grouping of materials properties, (called the “**PERFORMANCE INDEX**”), which maximise the objective

## Example 2: Performance Index for a light and stiff cylindrical tie

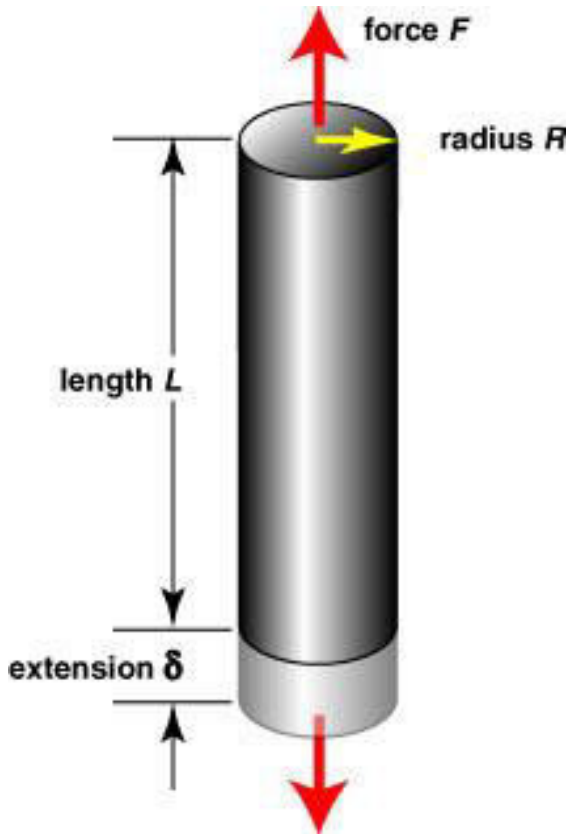


- A tie is loaded in **tension (FUNCTION)** (stiffness limited design at lowest mass)
- **Specified dimensions (CONSTRAINTS):**  $F$ ,  $L$  and  $\delta$ .
- **Free variable:** cross-sectional area,  $A$ .
- **Mass (OBJECTIVE):**

$$m = A l \rho \quad (1)$$

- **Deflection:**

$$\delta = L \varepsilon = L \frac{\sigma}{E} = L \frac{F}{AE} \quad (2)$$



- This gives the area  $A$ ;

$$A = \frac{FL}{\delta E} \quad (3)$$

- Substituting (3) into (1) gives:

$$m = \frac{FL}{E\delta} L\rho = \frac{FL^2}{\delta} \frac{\rho}{E} \quad (4)$$

- To minimise the weight (mass) the material performance index

$$M = \frac{E}{\rho} \quad (5)$$

should be maximised and the **E -  $\rho$  chart** will be used to select the optimum material

### Example 3: Performance Index for a light & strong cylindrical tie

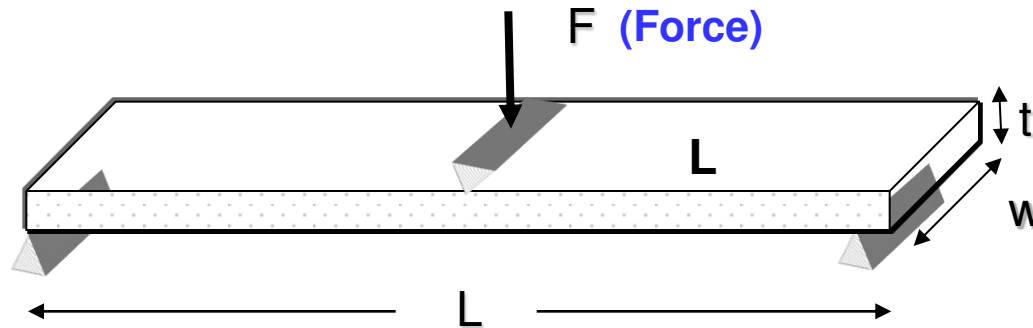
- A tie is loaded in tension (FUNCTION), carry load,  $F$  without failure
- Specified dimensions (CONSTRAINTS):  $F$ ,  $L$ , failure ( $\sigma_f$ )
- FREE VARIABLE: cross-sectional area,  $A$ .
- Reduce Mass (OBJECTIVE):

$$m = Al\rho \quad (1)$$

- The tie must carry load  $F$  without failure:  $\sigma_f = \frac{F}{A} \quad (2)$

$$m = \frac{F}{\sigma_f} l \rho \quad \longrightarrow \quad \boxed{M = \frac{\sigma_f}{\rho}}$$

#### Example 4: Performance Index for a light and stiff panel (fixed width)



##### FUNCTION

Panel (beam): bending

##### OBJECTIVE

Reduce weight:  $m = AL\rho = w t L \rho$  (1)

##### CONSTRAINT

Stiffness  $S = F / \delta = CEI / L^3$ ,  $I = w t^3 / 12$  (2)

##### FREE VARIABLE

Thickness,  $t$



We get the free variable,  $t$ , from Eq. (2),

$$S = \frac{CEI}{L^3} = \frac{CEwt^3}{12L^3} \quad \longrightarrow \quad t = \left( \frac{12SL^3}{CEw} \right)^{1/3}$$

Replacing  $t$  into Eq. 1, we get:

$$m = \left( \frac{12Sw^2}{C} \right)^{1/3} L^2 \left( \frac{\rho}{E^{1/3}} \right)$$

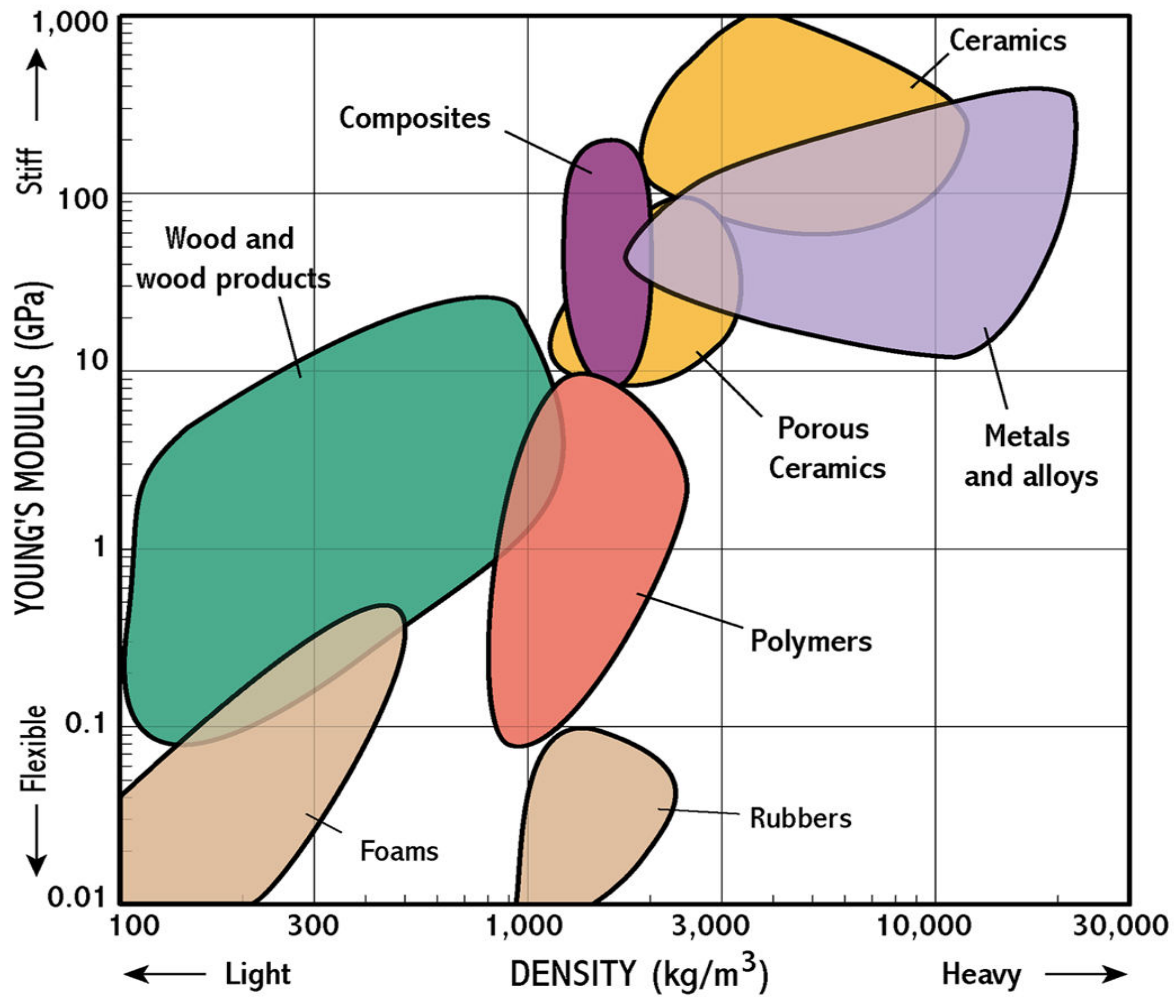
We select materials with the largest

$$M = \frac{E^{1/3}}{\rho}$$

Two concepts are used in the selection procedure:

## 2) Materials Selection Charts

- The selection of the optimum material is made more simple by the use of “*Materials Selection Charts*”
- Materials selection charts are plots of the properties that form the maximising factors.
- The performance index is made up of 2 properties (e.g;  $E$  &  $\rho$ ). The material selection chart is then created with axis (log by default) of  $E$  and  $\rho$ .

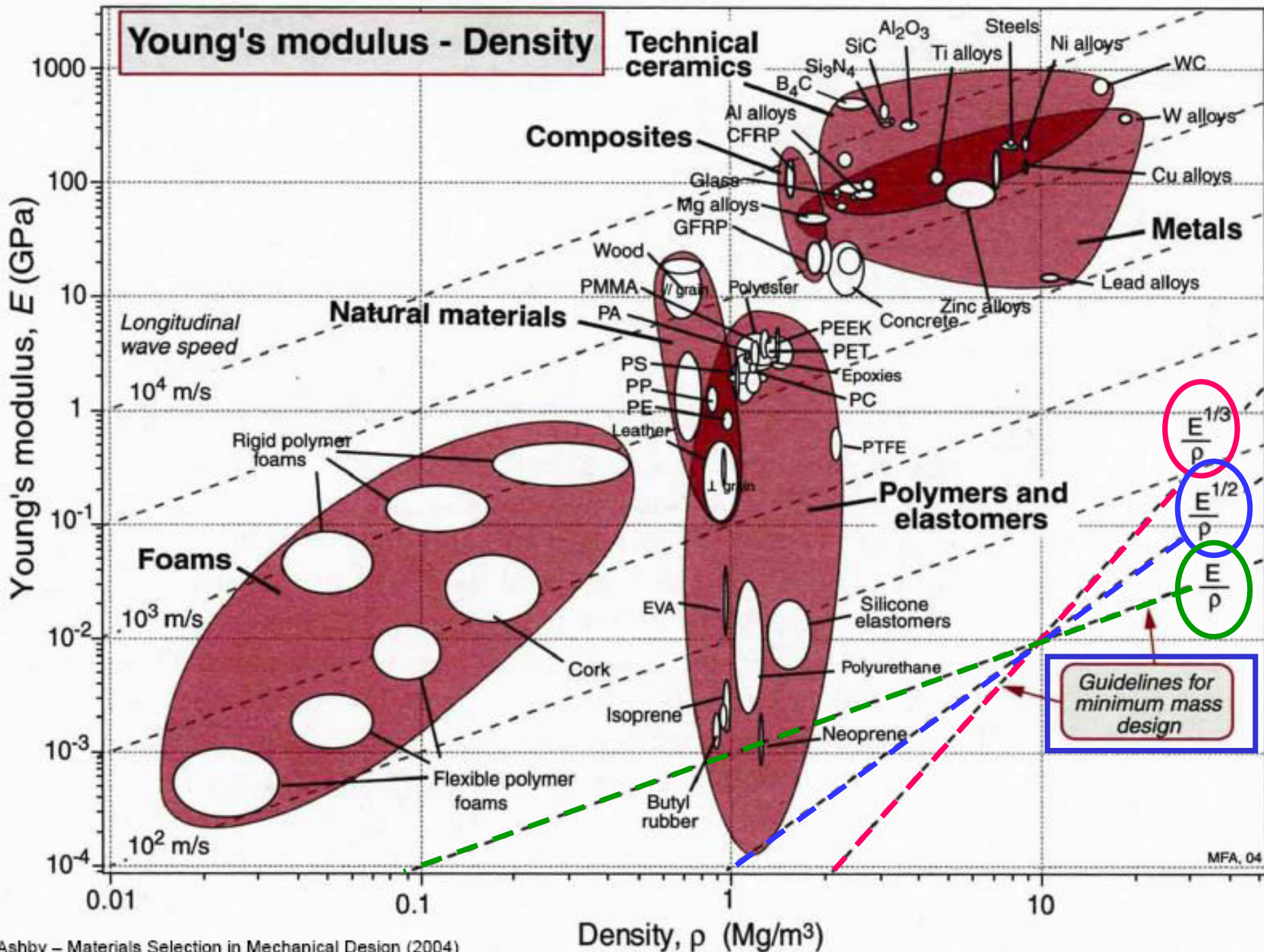


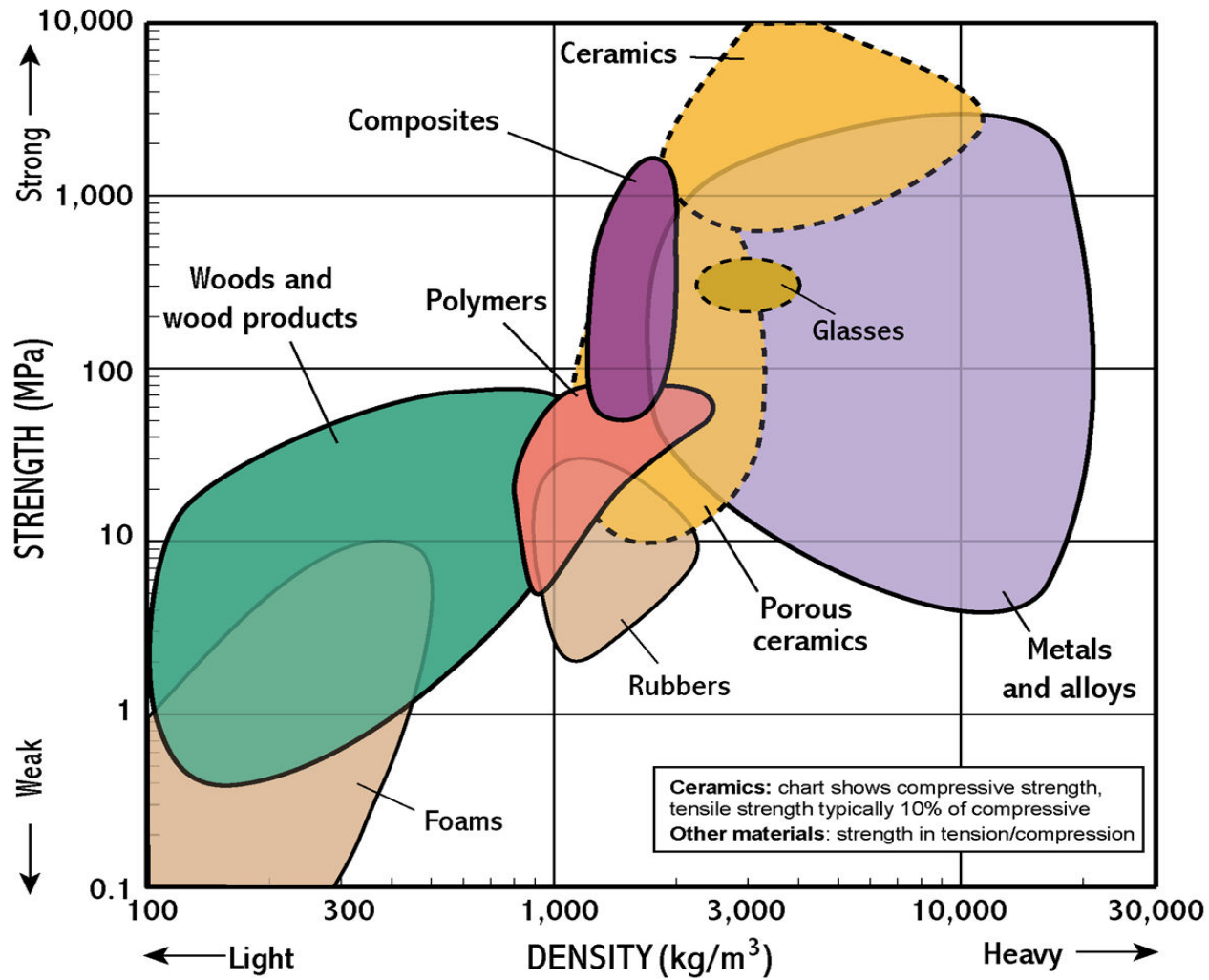
After M.F. Ashby

- The performance index,  $M$ , plots as a diagonal line on the chart. Its slope is very important.
- For the performance index  $M = E / \rho$ . We take log;

$$\text{Log } E = \log M + \log \rho$$

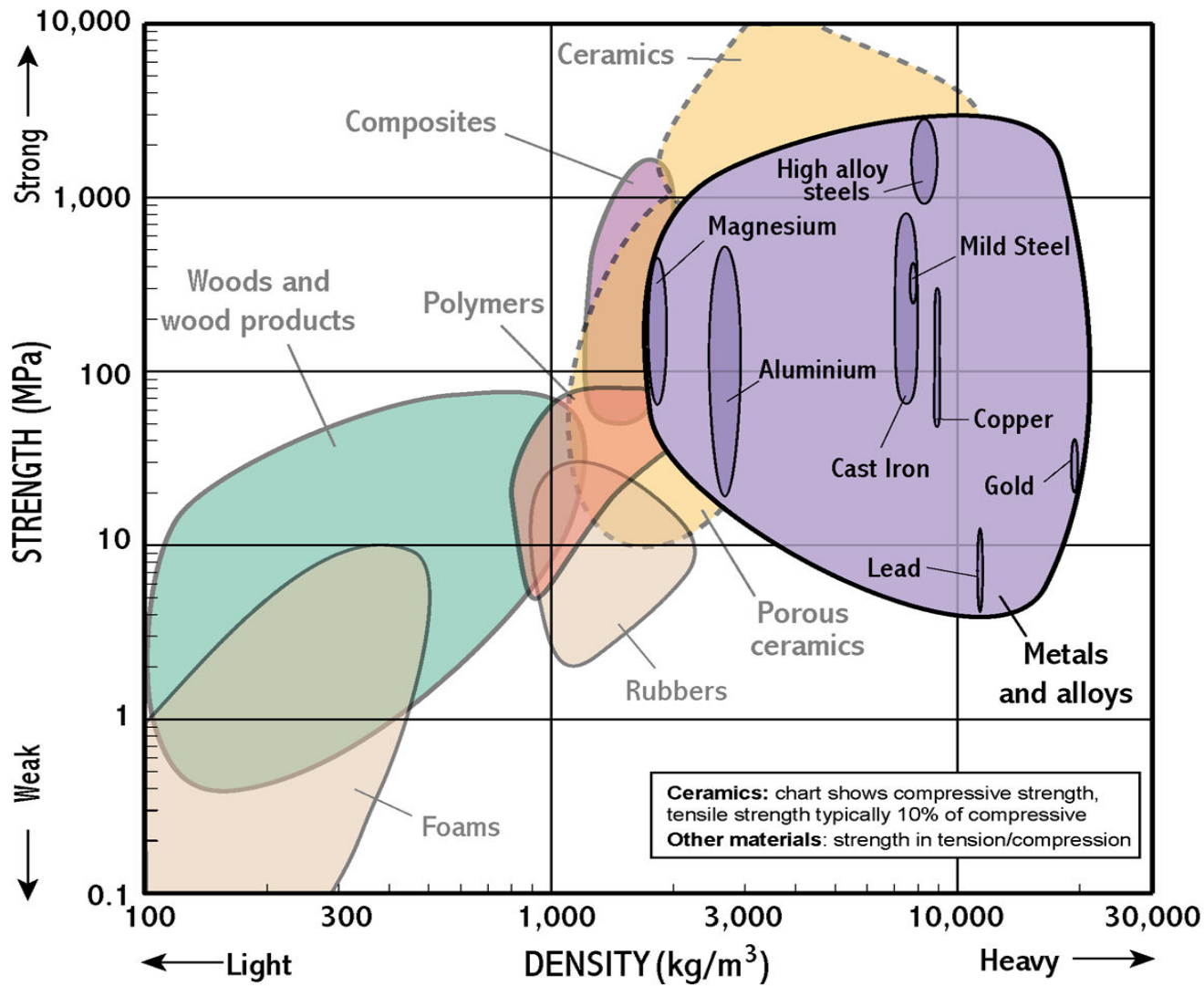
- A line of slope 1 on the chart describes the index; its position is determined by the value of  $M$ .
- For  $M = E^{1/2} / \rho$  and  $E^{1/3} / \rho$ , gives lines with slope 2 and 3 respectively.
- They are called “*design guidelines*”





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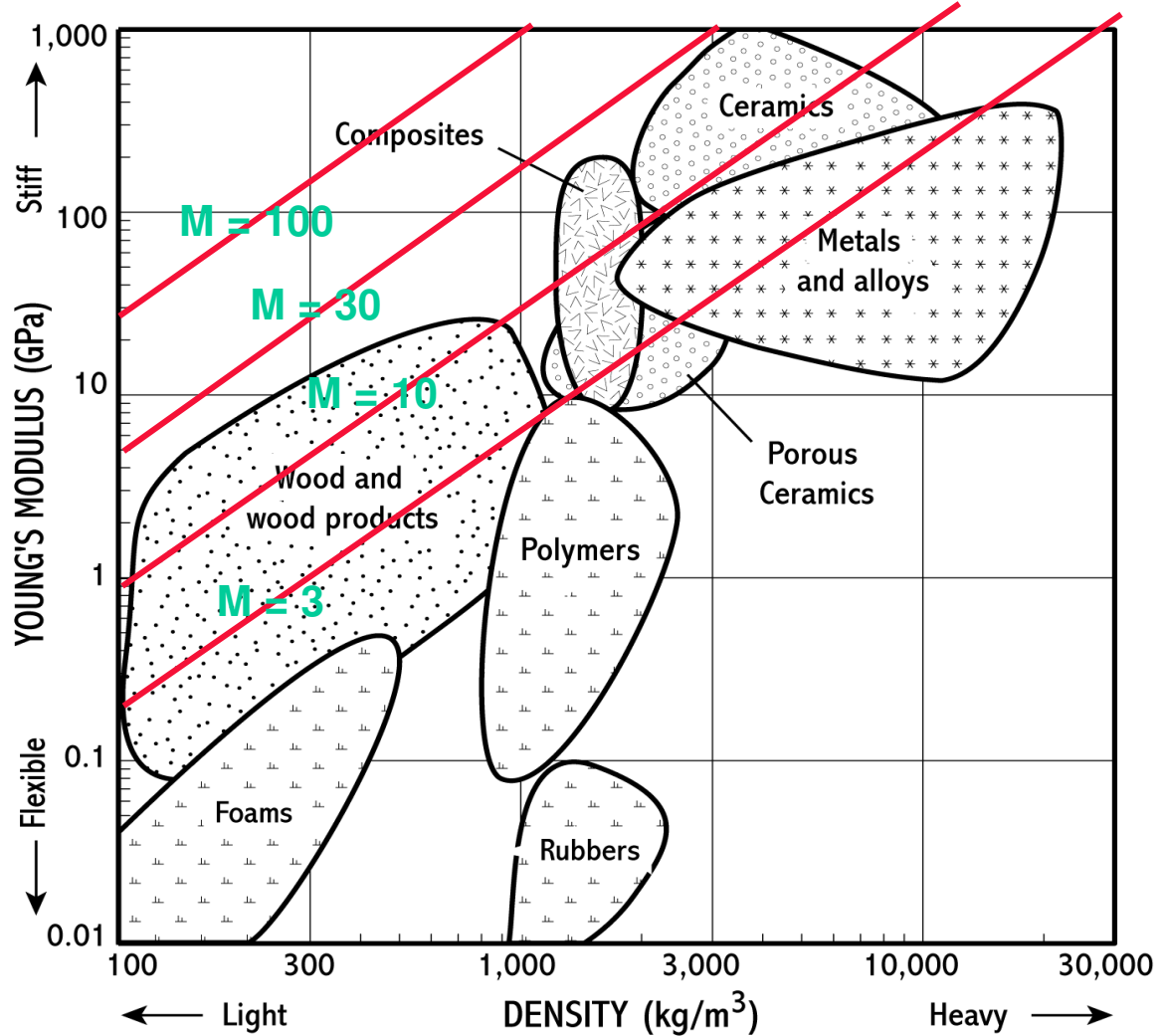


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All parallel lines  
have the same  
performance

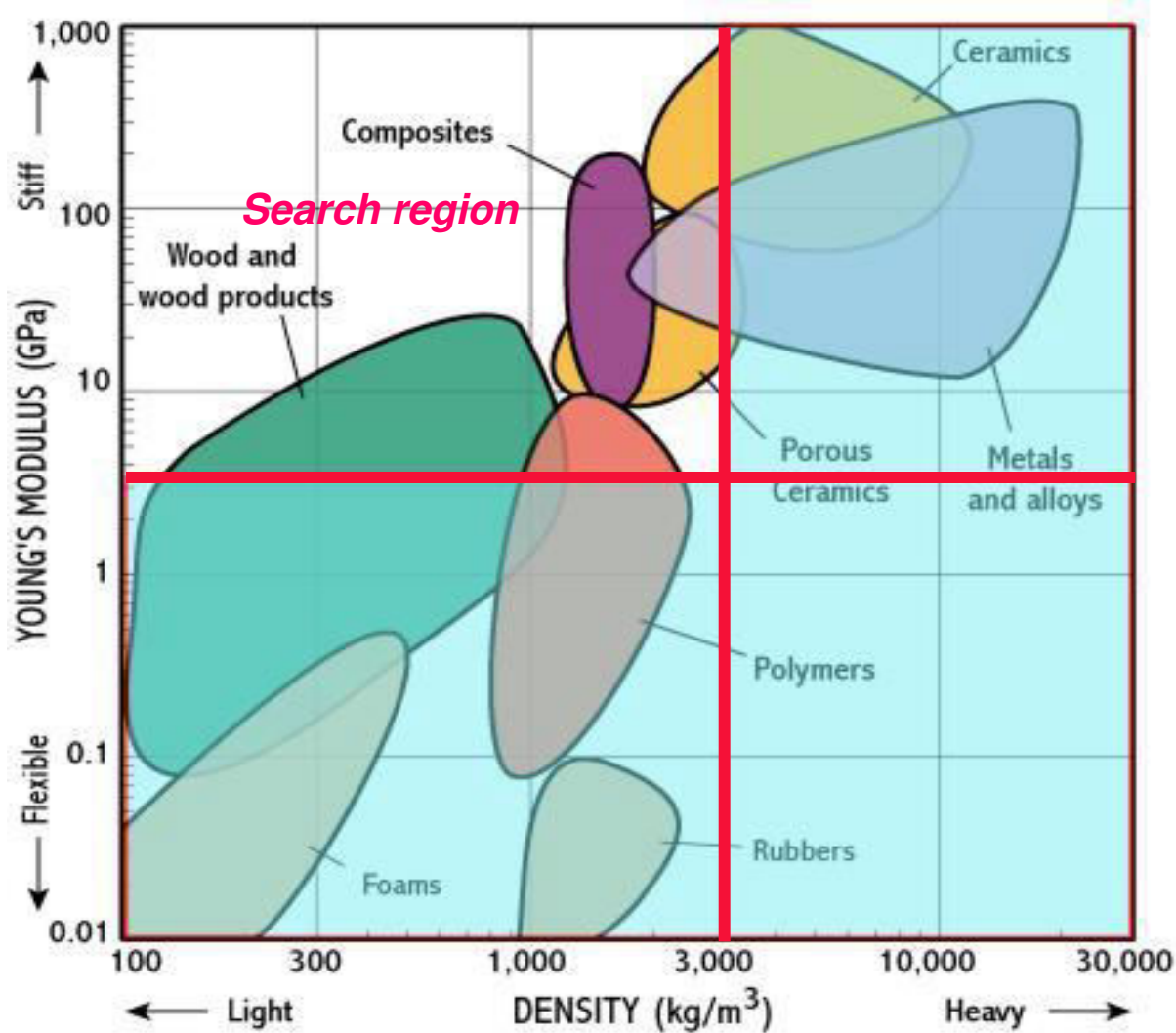
$M = 100$  is 1/10 the  
weight of  $M = 10$

$M = 30$  is 1/3 the  
weight of  $M = 10$



## Constraints on the materials selection charts

- Selection of a material is influenced by constraints.
- Constraints appear as *horizontal* or *vertical lines* on the selection chart
- The primary constraints eliminate blocks of materials leaving a *viable search region*.
- The next step in selecting a material operates only on those materials which are left inside the search region

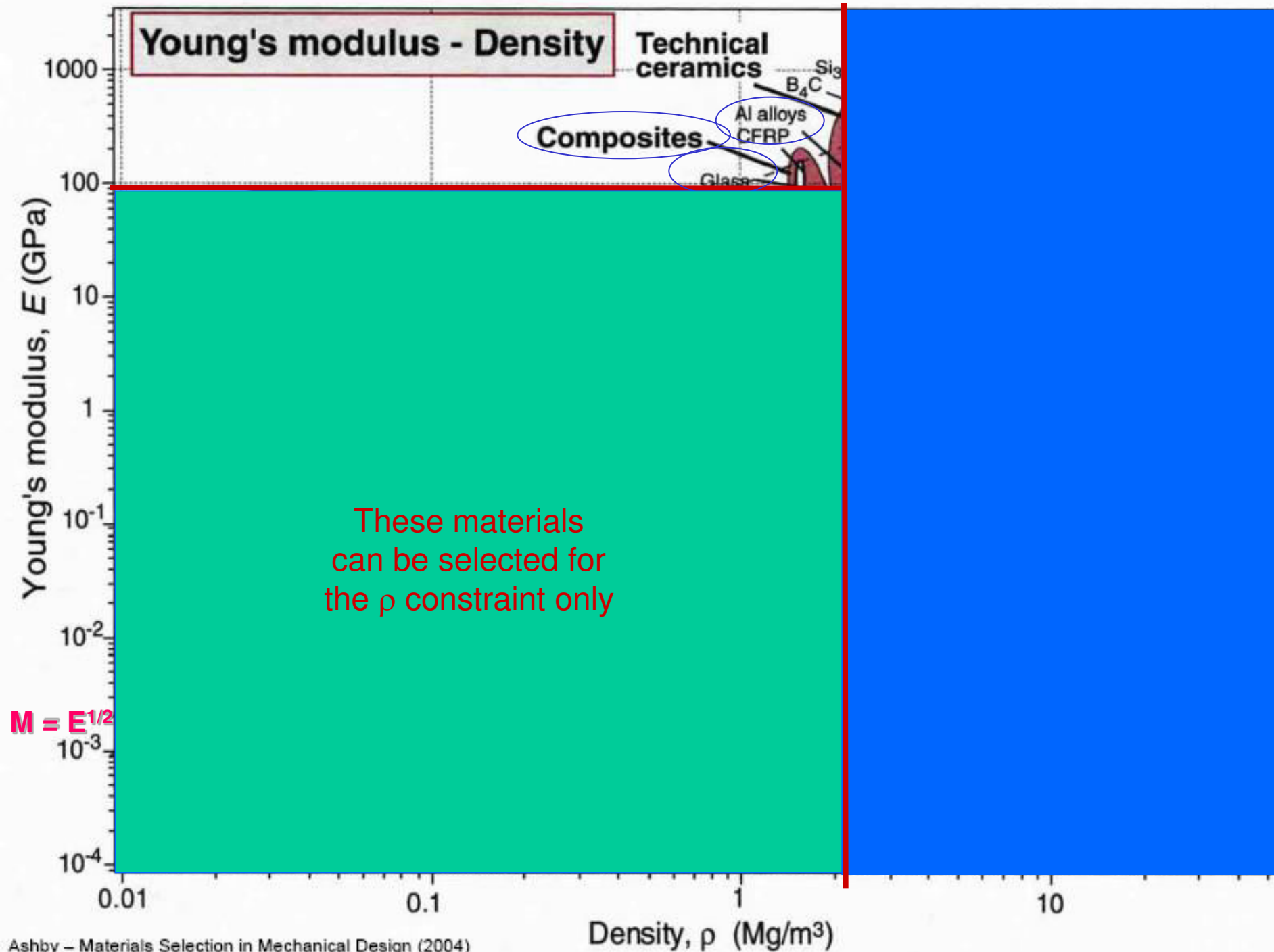


Red lines are  
Constraints

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## *Example*

Use the E-  $\rho$  chart to find the material with modulus  $E > 90$  GPa and density  $\rho < 2$  Mg/m<sup>3</sup>



## Multiple Constraints

- Most materials selection problems are over-constrained with more constraints than free variables.
- e.g; a component may require minimum weight, but with constraints on stiffness, strength and toughness which must also be met.
- This requires the use of several performance indices and several materials selection charts to identify the optimum material

- To solve this problem the following steps are considered:
  1. Identify the most important constraint [minimum weight without yielding (plastic deformation)] and identify the appropriate materials performance index. Ignore the remaining constraints.
  2. Use this to identify a subset of materials which maximise this performance index.
  3. Repeat for the next constraint(s) giving a second or a third performance index.

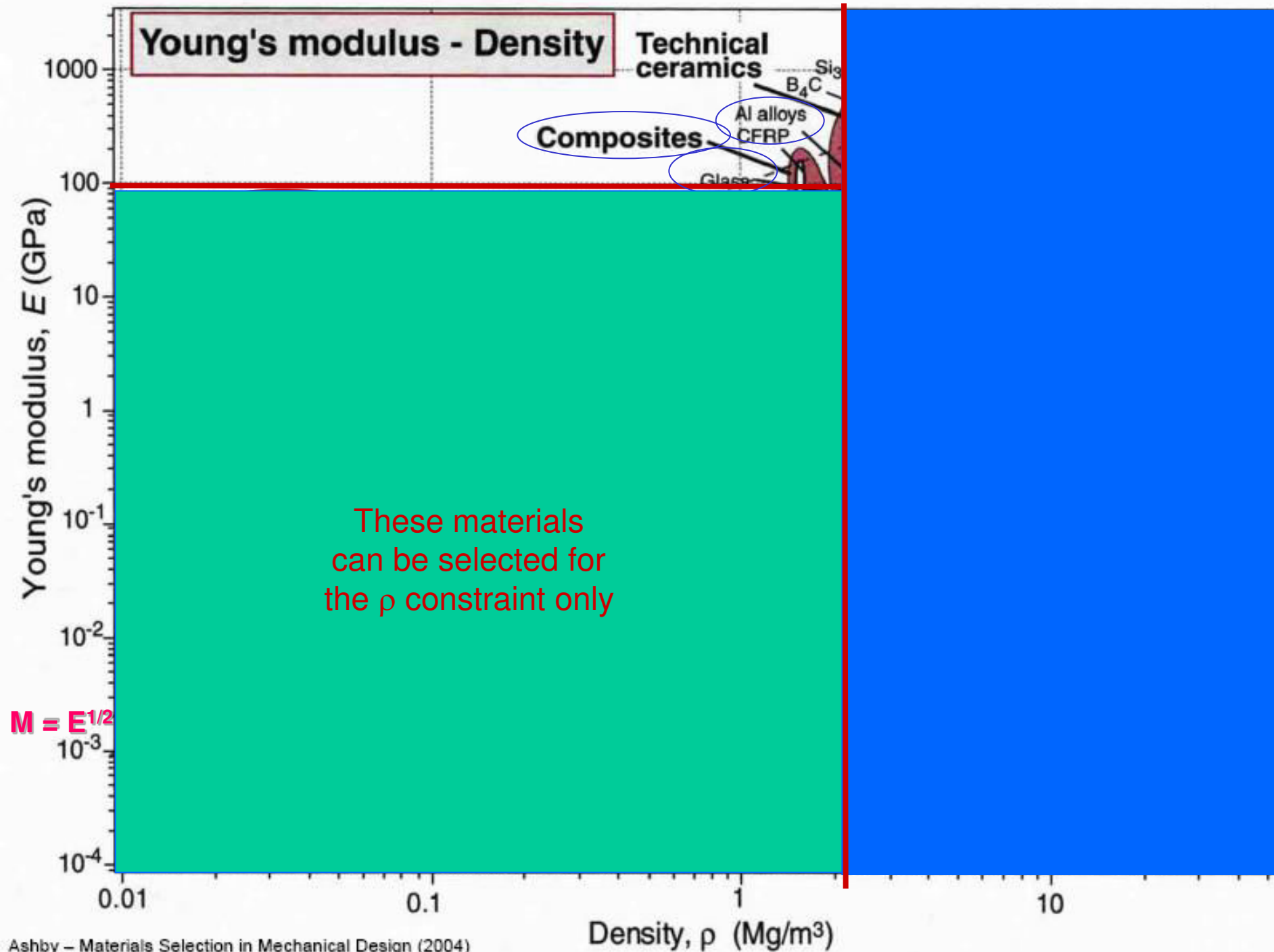


4. Then, identify a subset of materials which satisfy all performance indices.

- The second index may be plotted on the same chart as the first. The sector isolated above the two lines contains the subset of materials which satisfy the two criteria:  $E^{1/2} / \rho > 8 \text{ GPa}^{1/2} / \text{Mg/m}^3$  and  $E > 10 \text{ GPa}$ .
- More often, however, the extra constraint involves a property which does not appear on the first chart.
- In this case, the members of the first subset of materials are tabulated, ranking them using a grid of performance index values

## Example

1. Use the E-  $\rho$  chart to find the material with modulus  $E > 100 \text{ GPa}$  and density  $\rho < 2 \text{ Mg/m}^3$
2. Use the E-  $\rho$  chart to identify the subset of materials with both modulus  $E > 100 \text{ GPa}$  and the performance index  $M = E^{1/2} / \rho > (6 \text{ GPa})^{1/2} / (\text{Mg/ m}^3)$



# Young's modulus - Density

Technical ceramics

Composites

Al<sub>2</sub>O<sub>3</sub>  
SiC  
Si<sub>3</sub>N<sub>4</sub>  
B<sub>4</sub>C  
Al alloys  
CFRP  
Glass  
Ti alloys  
Steel

These materials  
can be selected for  
the  $M=E^{1/2}/\rho$   
constraint only

$$M = E^{1/2}/\rho$$

Density,  $\rho$  (Mg/m<sup>3</sup>)

- The second index is used, with the appropriate chart, to identify a second subset of materials. Common members of the two subsets are identified and ranked according to their success in maximising the two performance indices.
- Consider a problem with one design goal, one free variable and two constraints.
- The result is two equations for the performance, each with the equation:

$$P = f_1 (F) \cdot f_2 (G) \cdot f_3 (M)$$

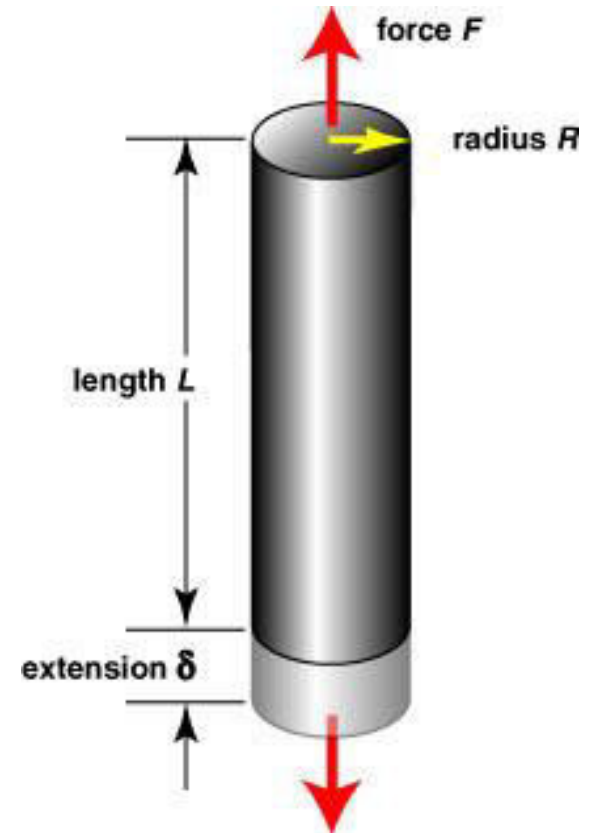
$$P = g_1 (F) \cdot g_2 (G) \cdot g_3 (M)$$

- The performance is maximised by choosing:
  1. Material with the largest “ $f_3(M)$ ”
  2. Material with the largest “ $g_3(M)$ ”
- The two equations must have equal values. The two performance indices are coupled.

$$\frac{f_3(M)}{g_3(M)} = \frac{g_1(F) \times g_2(G)}{f_1(F) \times f_2(G)}$$

### Example 4: Performance Index for a light, stiff and strong tie

- The tie (loaded in tension) is to support a load  $F$ , at minimum weight, without failing or extending by more than  $\delta$
- Objective function:* reduce weight
$$m = A L \rho \quad (1)$$
- Specified dimensions:  $F$  and  $L$
- Constraints: failure and stiffness
- Free variable: cross-sectional area,  $A$ .



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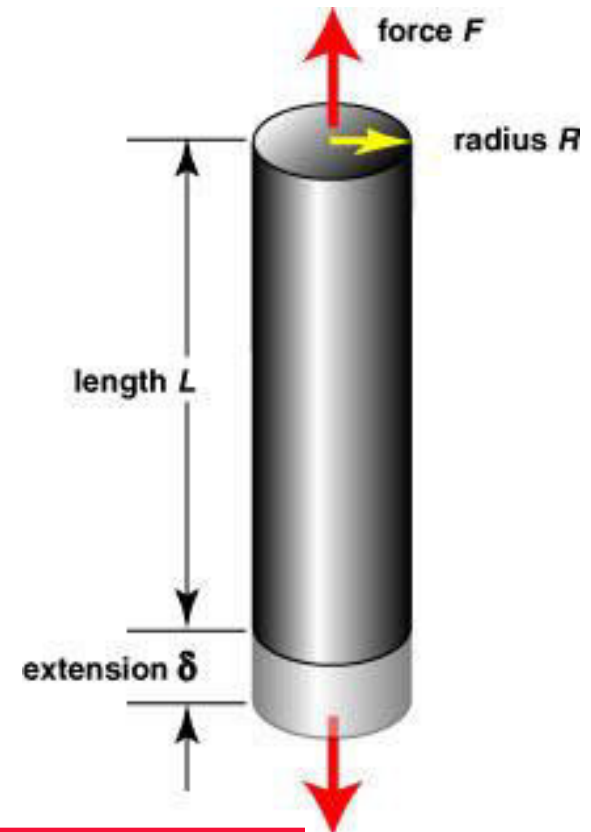
- We need to derive 2 performance indices
- $M_1$  for the stiffness constraint has already been derived in example 2 as:

$$M_1 = \frac{E}{\rho}$$

- $M_2$  for the failure constraint has also been derived in example 2 as:

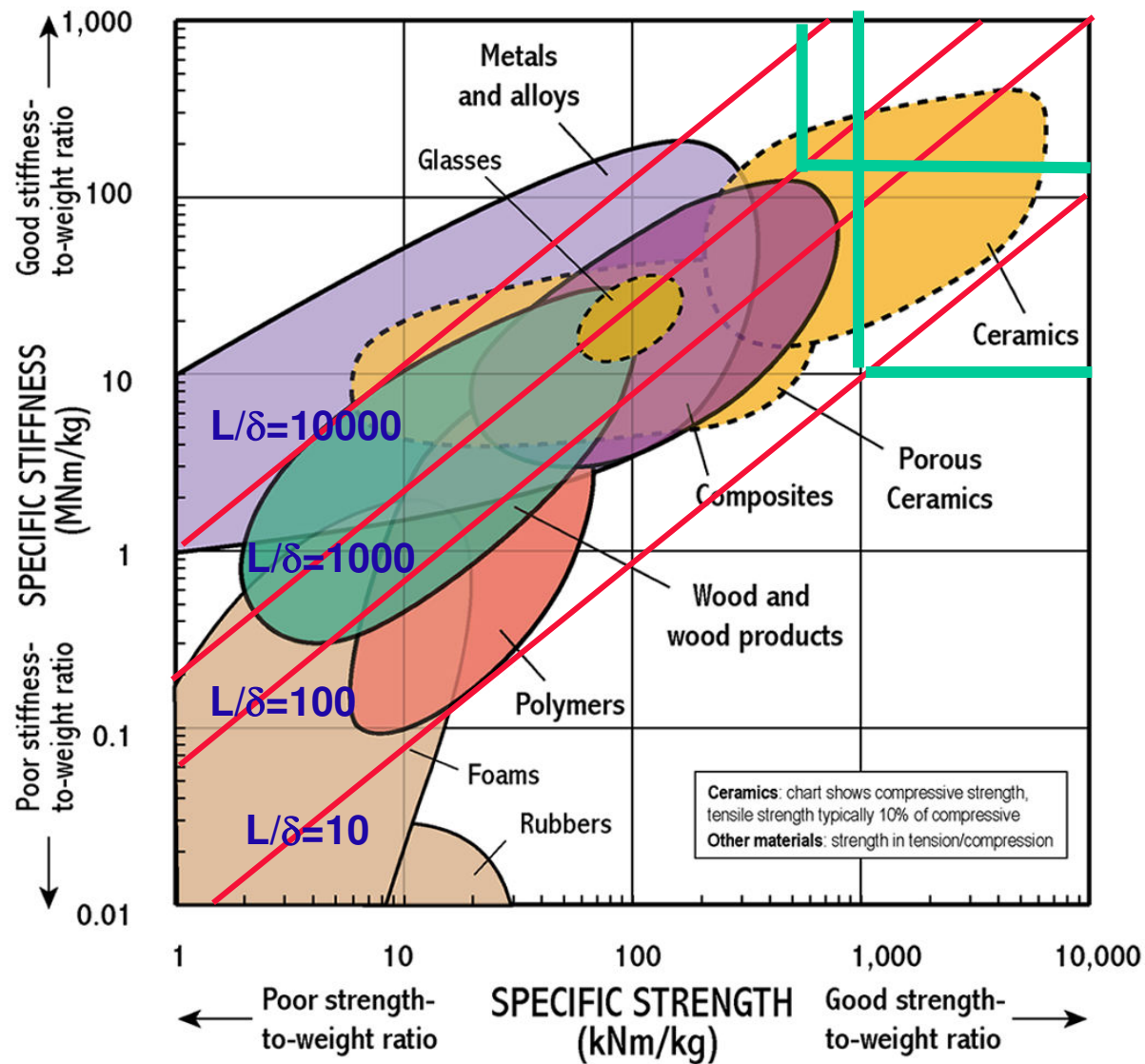
$$M_2 = \frac{\sigma_f}{\rho}$$

- Since the weight is the same so that:  $M_1 = M_2$   
This is called the “*coupling equation*”



$$\frac{E/\rho}{\sigma_f/\rho} = \frac{L}{\delta}$$





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## Multiple Design Goals

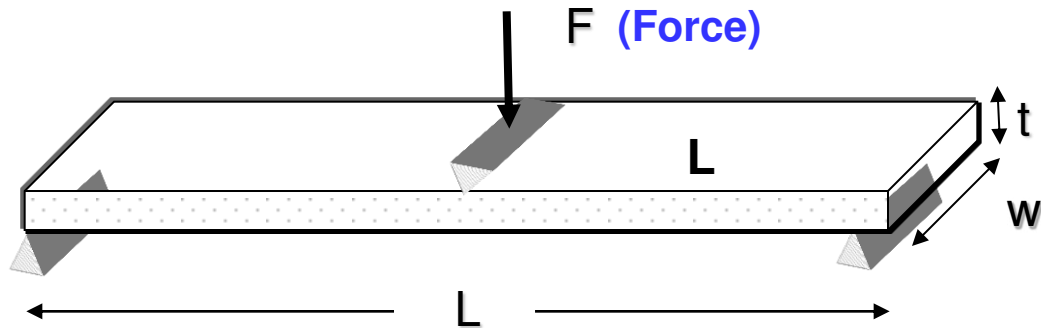
- For example, a design goal is to minimise weight, another is to minimise cost (how is weight to be compared to cost (they have different units?))
- The designer must assess the relative importance of all design goals by using “*weighting factors*” to each design goal
- First, the design goals are *ranked in order of importance*: a numerical factor of “10” is given to the most important goal and a factor of “1” is given to the least important

- The **performance index for each design goal** (starting with the most important) is determined
- Finally, the overall performance index is calculated by combining the performance index  $M$  obtained from each design goal
- Where  $a_1 > a_2 > a_3$ , are the weighting factors.

$$M = a_1M_1 + a_2M_2 + a_3M_3 + \dots$$

- Alternatively a trade-off between the objectives should be used. This is illustrated in the next example.

Example 5: Stiff electronic casing (notebook) with minimum thickness and weight



FUNCTION

Bending

OBJECTIVES

1. Minimum thickness,  $t$
2. Reduce weight:  $m = AL\rho = w t L \rho$

CONSTRAINT

Stiffness  $S = F / \delta = CEI / L^3$ ,  $I = w t^3 / 12$

FREE VARIABLE

$t$

See example 3.1.5

**OBJECTIVE 1** Minimum thickness,  $t$

Using the constraint equation we get,  $t$ , as:

$$t = \left( \frac{12SL^3}{CEw} \right)^{1/3} \rightarrow M_1 = E^{1/3}$$

**OBJECTIVE 2** Reduce weight,  $m$

Using  $m = w t L \rho$  and replacing  $t$  into the mass equation we get:

$$m = AL\rho = \left( \frac{12Sw^2}{C} \right)^{1/3} L^2 \left( \frac{\rho}{E^{1/3}} \right) \rightarrow M_2 = \frac{E^{1/3}}{\rho}$$

- For multiple objectives, we need to determine the relative performance indices.
- That is, suppose the casing is currently made of material  $M_0$

1. The thickness of a casing made from an alternative material  $M$ , will be different from that made of material  $M_0$  by the factor:

$$\frac{t}{t_0} = \left( \frac{E_0}{E} \right)^{1/3}$$

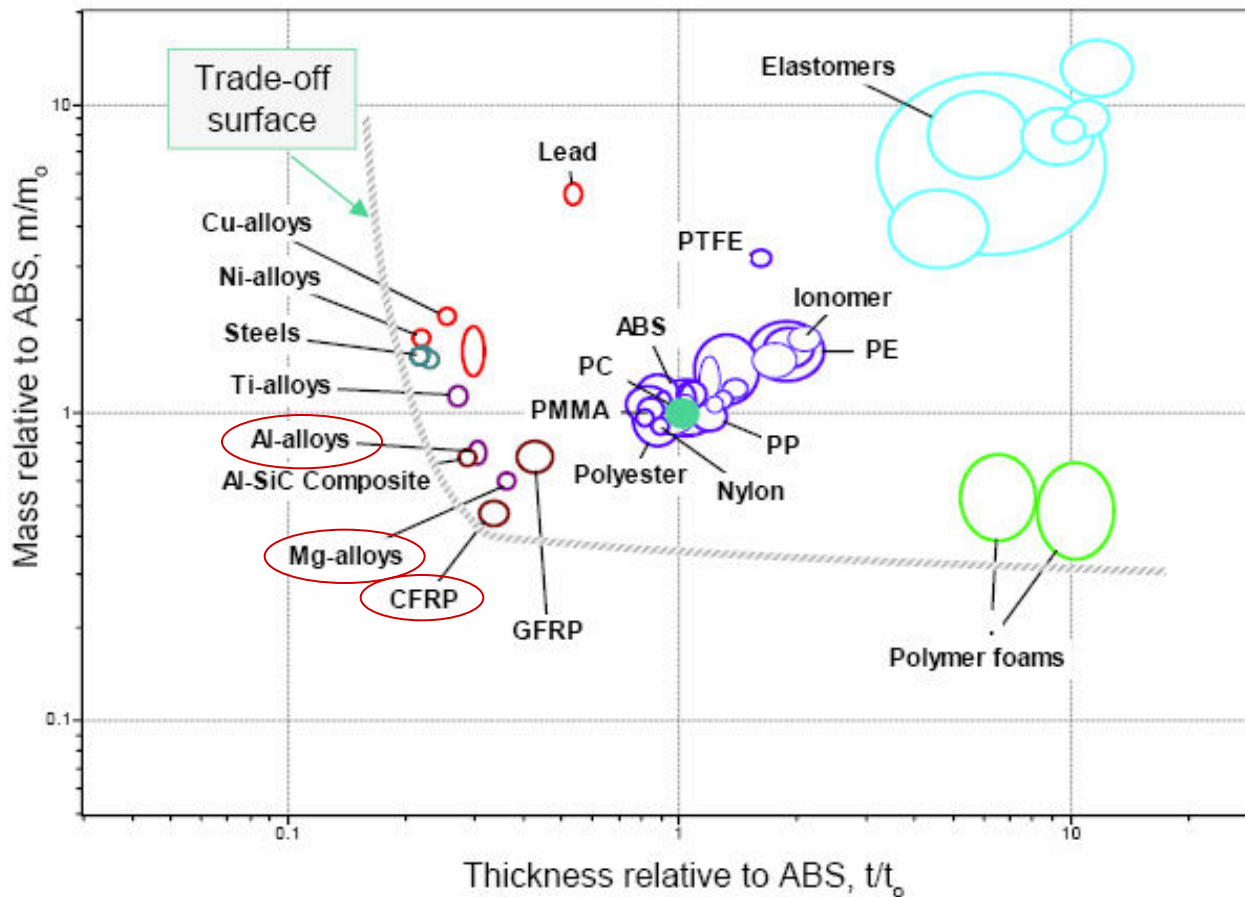
2. The mass of a casing made from an alternative material  $M$ , will be different from that made of material  $M_0$  by the factor:

$$\frac{m}{m_0} = \left( \frac{\rho}{E^{1/3}} \right) \left( \frac{E_0^{1/3}}{\rho_0} \right)$$

- The trade-off between thickness and mass can be determined using the chart

$$\frac{t}{t_0} \quad \text{and} \quad \frac{m}{m_0}$$

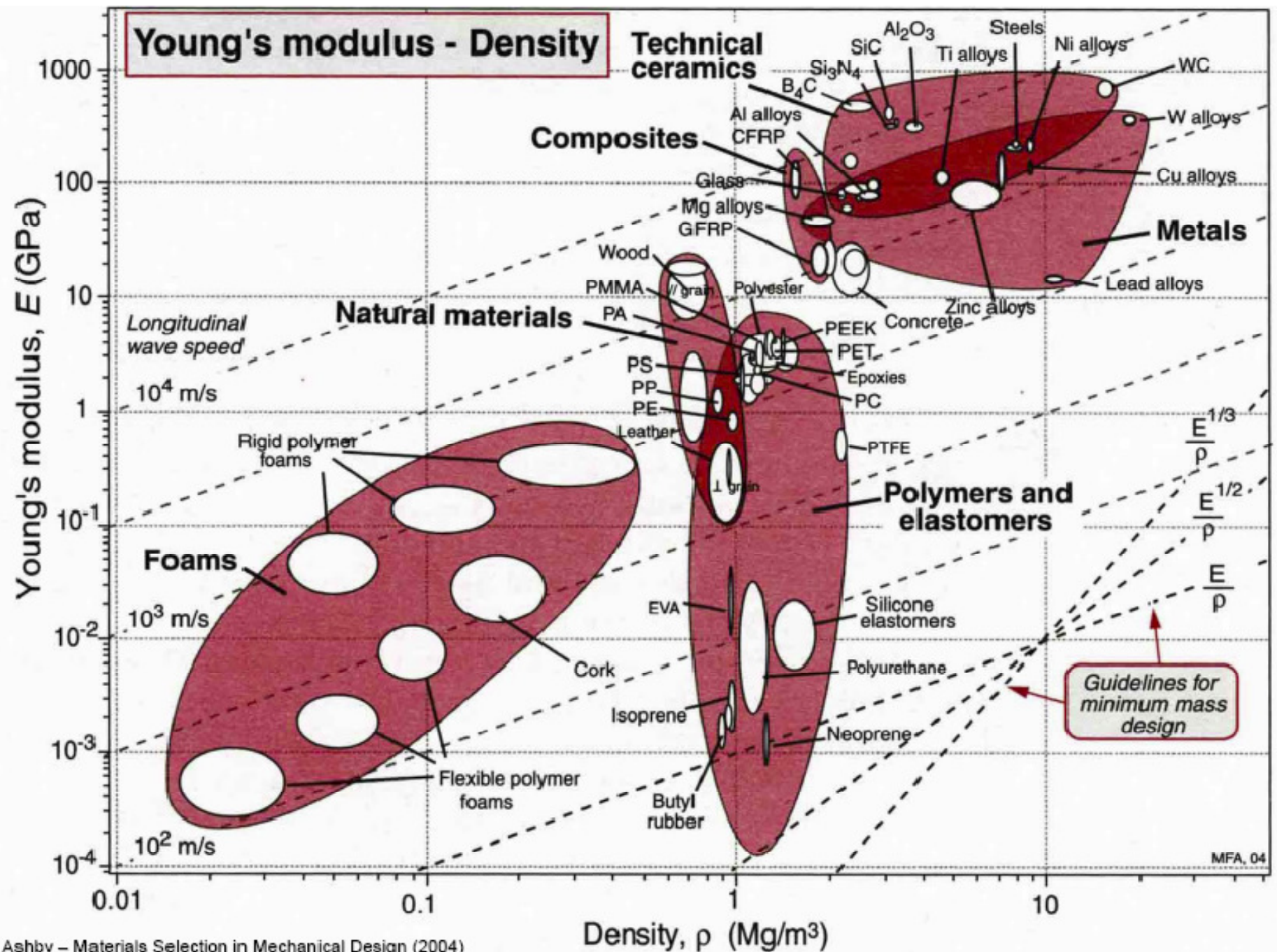
## TRADE-OFF PLOT

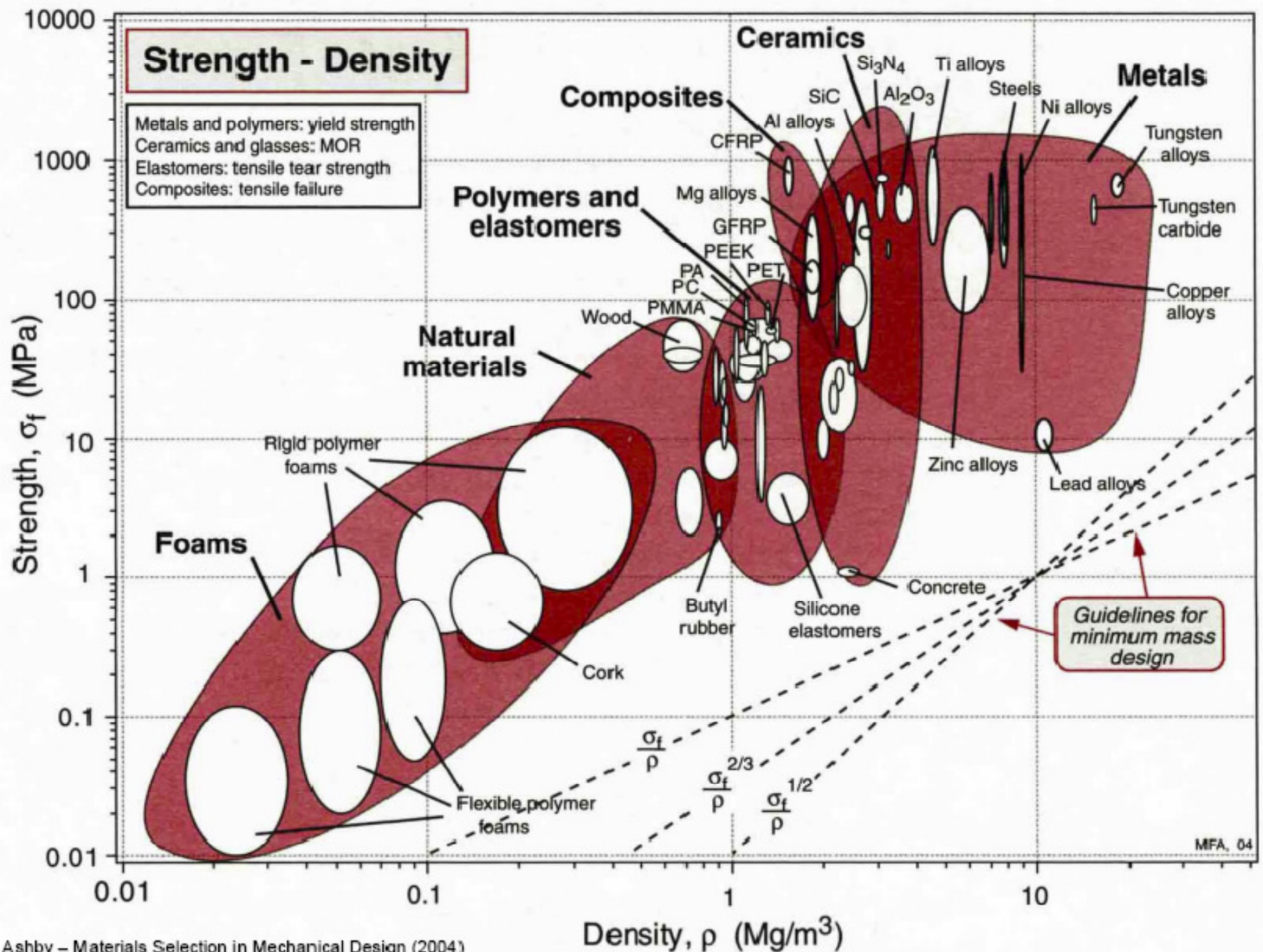


**POSSIBLE MATERIALS: CFRP, Al & Mg alloys (Offer low mass at minimum thickness)**

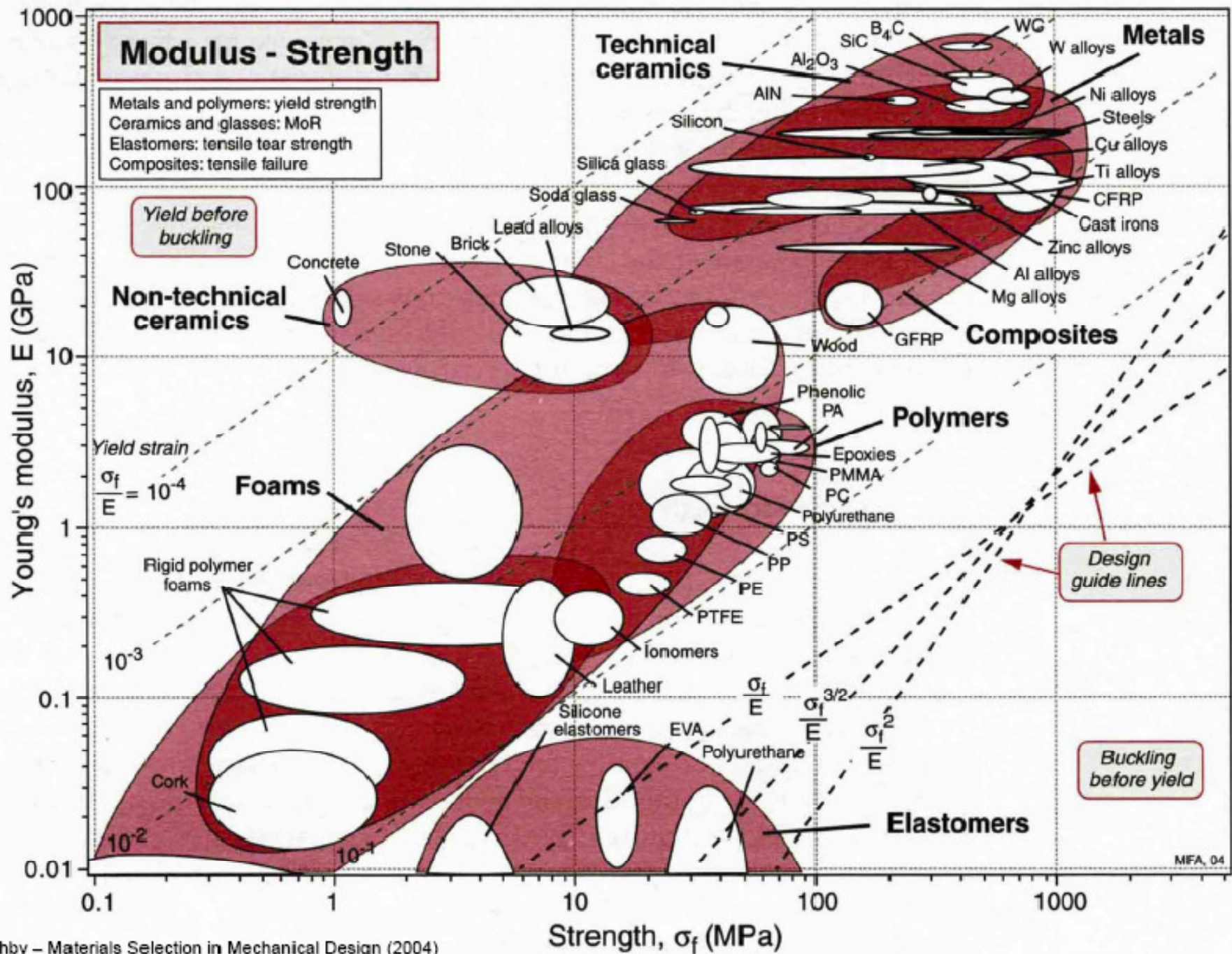
# MATERIALS SELECTION CHARTS



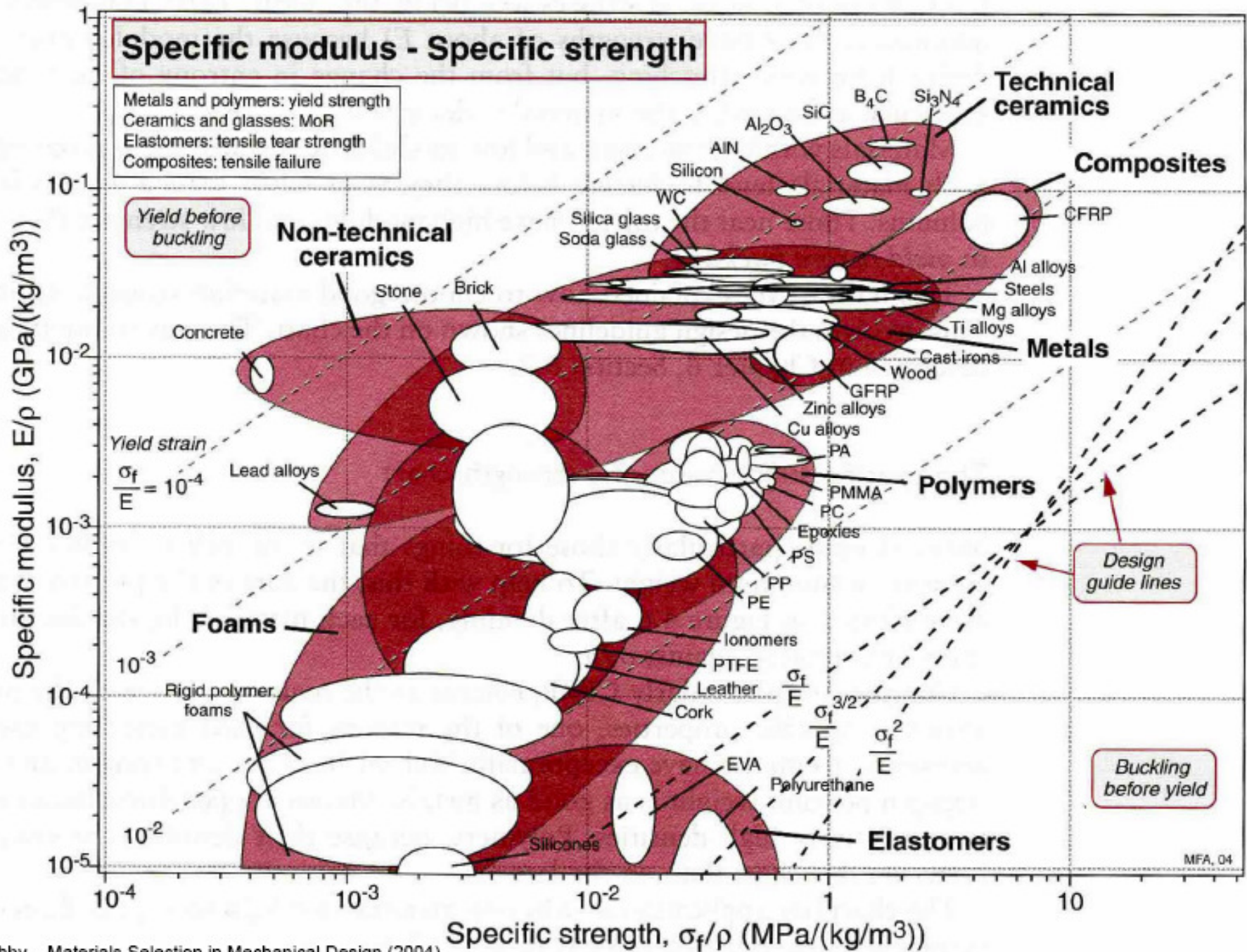




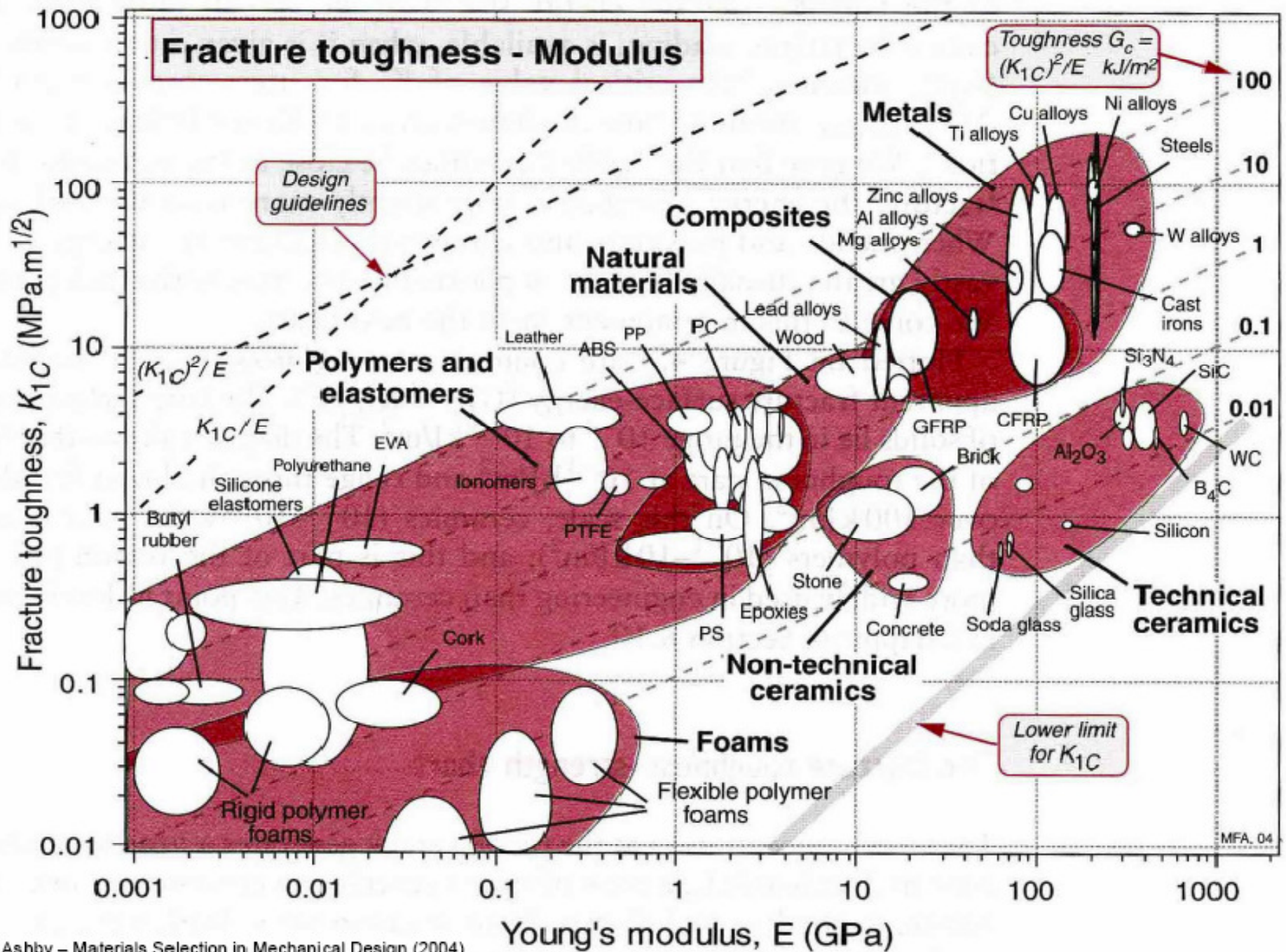




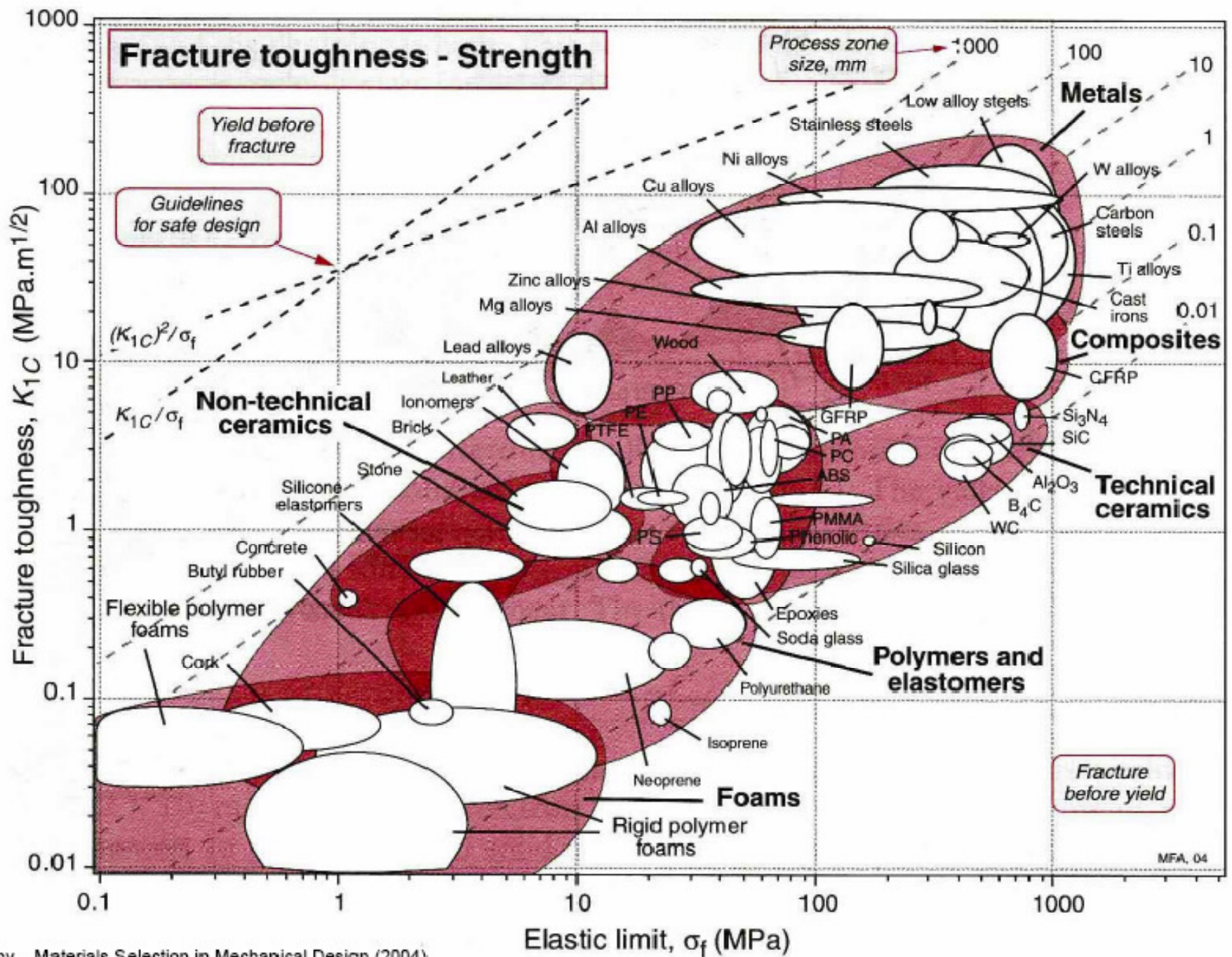




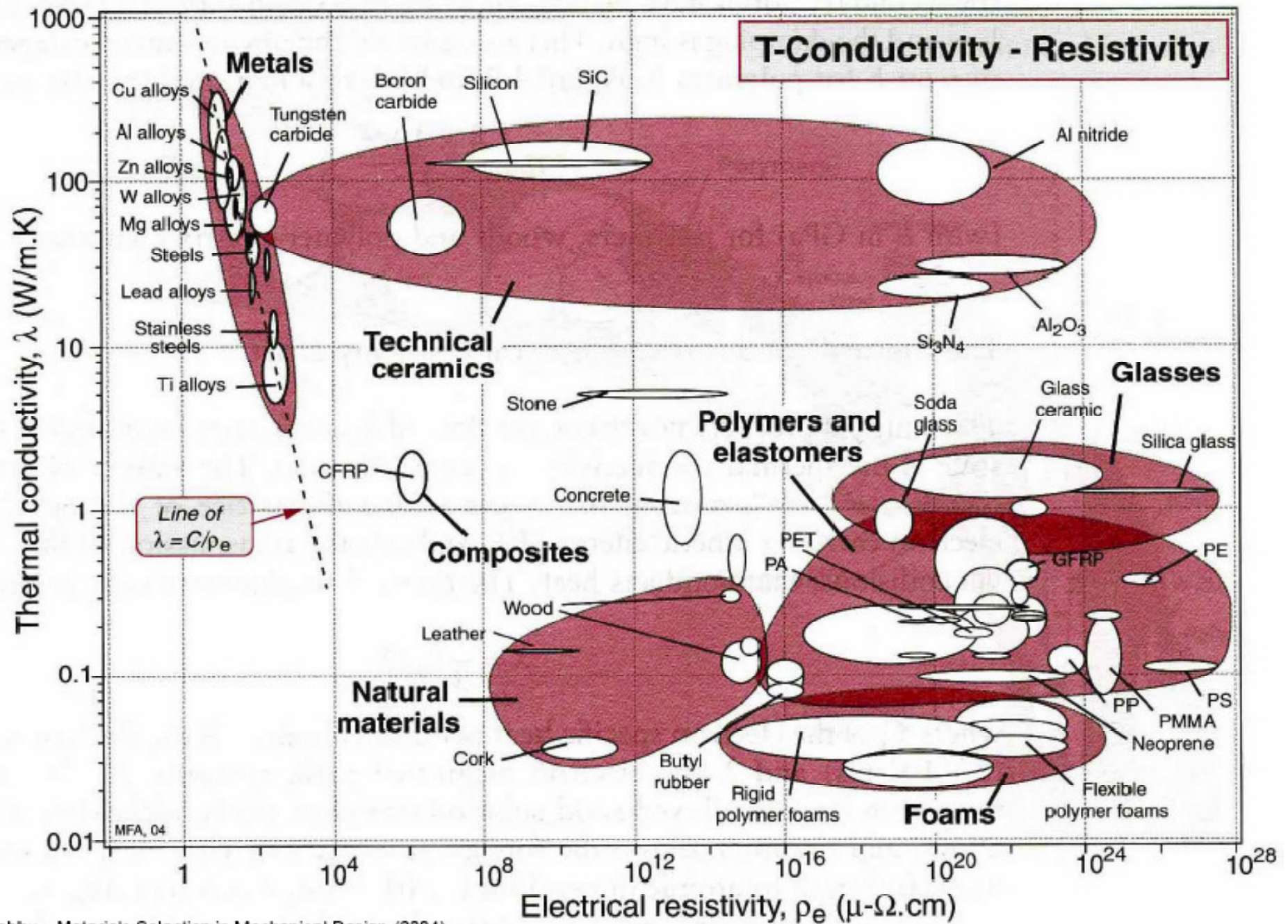








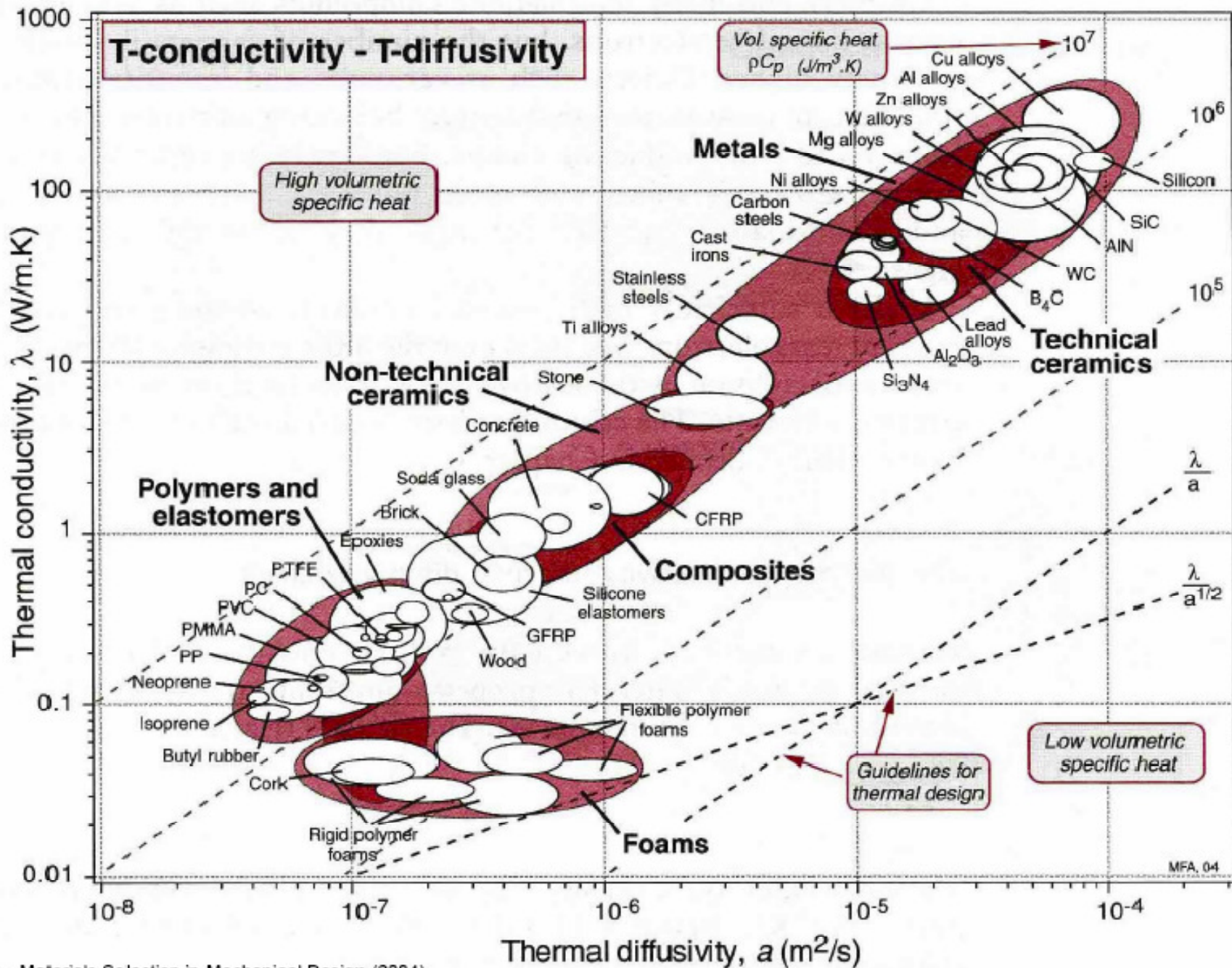


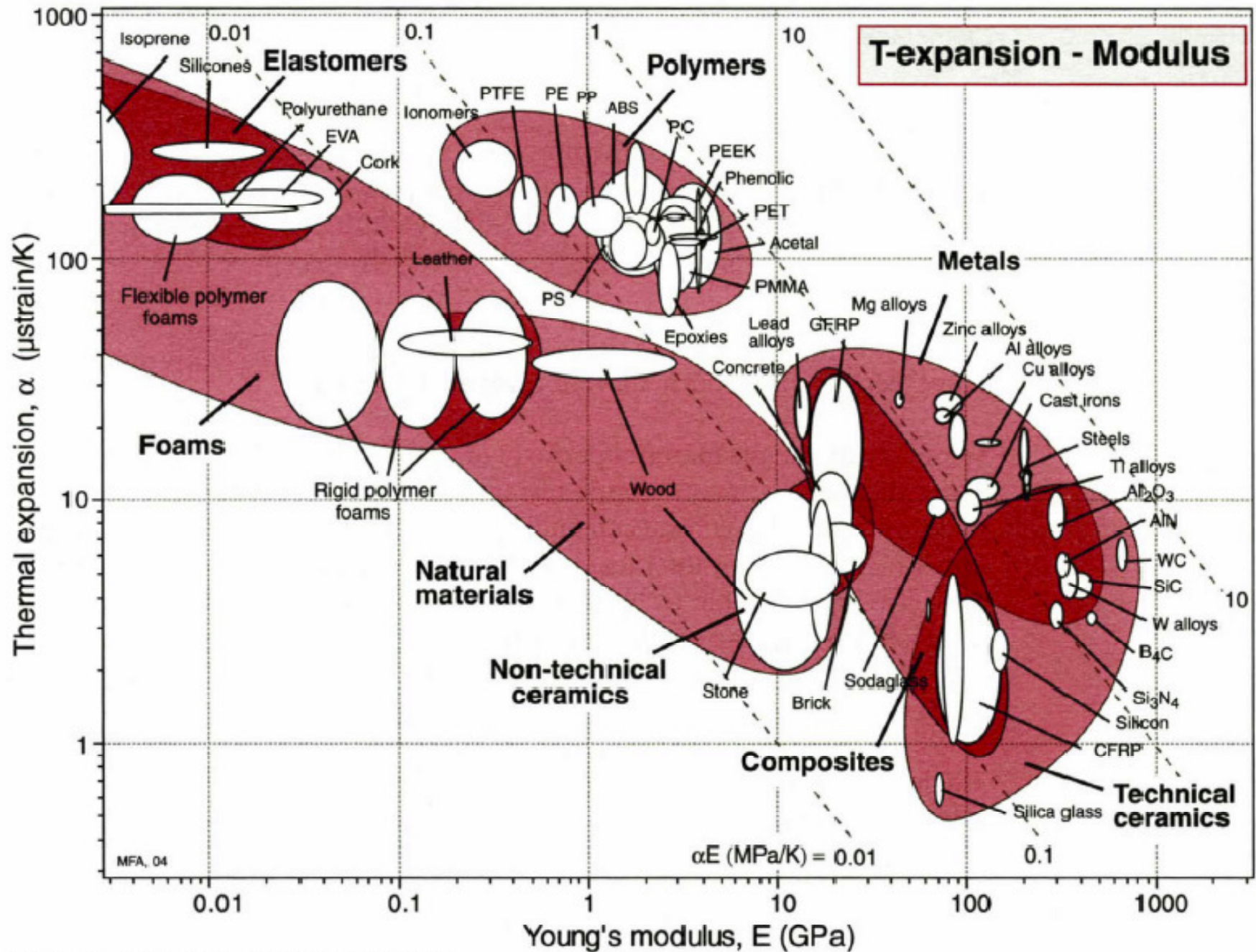




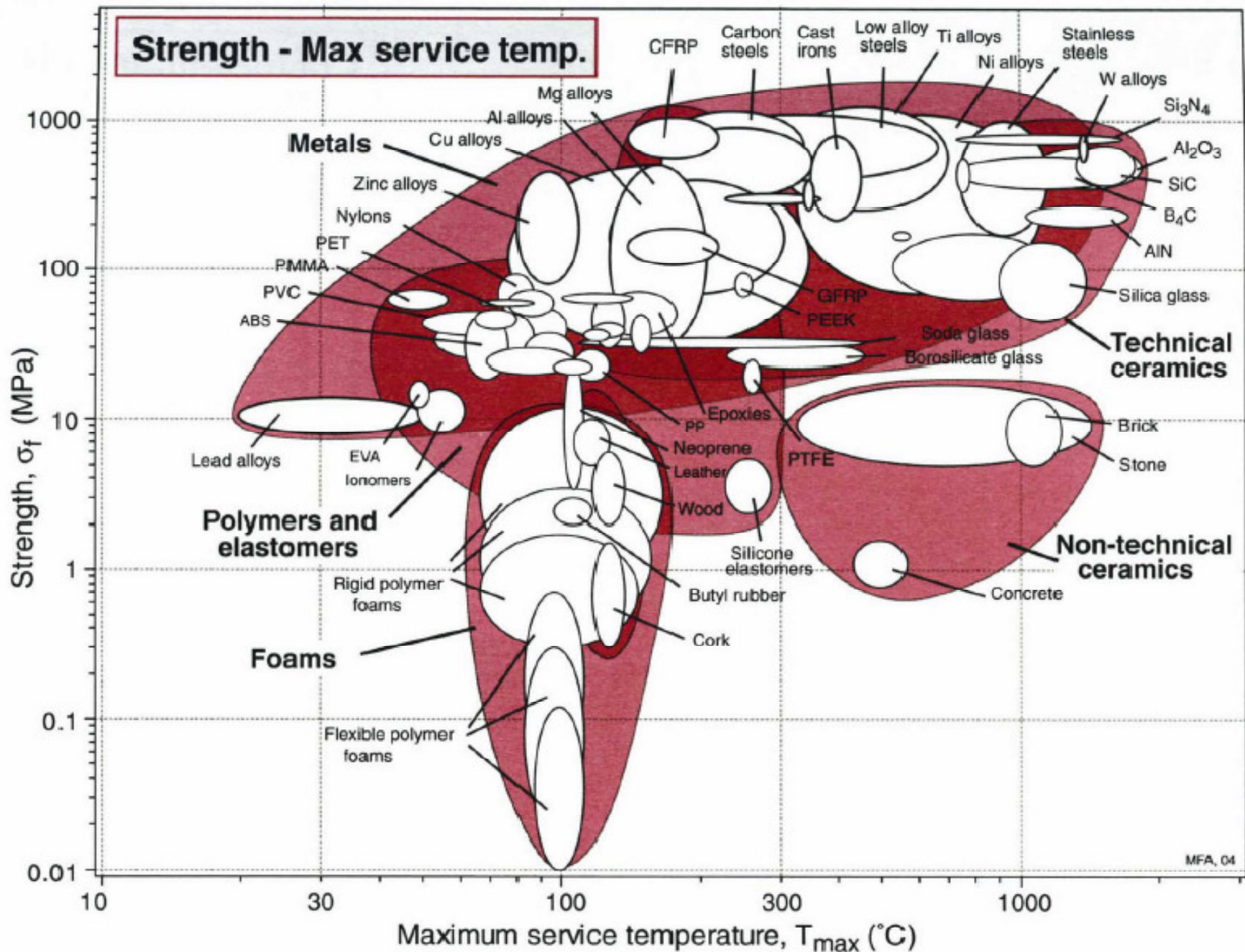












MFA, 04

